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Data-Analytics-Driven Selection of Die Material in Multi-Material Co-Extrusion of Ti-Mg Alloys

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Abstract: The selection of the most suitable material is one of the key decisions to be made during the design stage of a manufacturing process. Traditional approaches, such as Ashby maps based on material properties, are widely used in industry. However, in the production of multi-material components, the criteria for the selection can include antagonistic approaches. The aim of this work is to implement a methodology based on the results of process simulations for several materials and to classify them by applying an advanced data analytics method based on machine learning (ML)—in this case, the support vector regression (SVR) or multi-criteria decision-making (MCDM) methodology. Specifically, the multi-criteria optimization and compromise solution (VIKOR) was combined with entropy weighting methods. To achieve this, a finite element model (FEM) was built to evaluate the extrusion force and the die wear during the multi-material co-extrusion process of bimetallic Ti6Al4V-AZ31B billets. After applying SVR and VIKOR in combination with the entropy weighting methodology, a comparison was established based on material selection and the complexity of the methodology used. The results show that the material chosen in both methodologies is very similar, but the MCDM method is easier to implement because there is no need for evaluating the error of the prediction model, and the time required for data preprocessing is less than the time needed when applying SVR. This new methodology is proven to be effective as an alternative to traditional approaches and is aligned with the new trends in industry based on simulation and data analytics.



Citation: Fernández, D.; Rodríguez-Prieto, Á.; Camacho, A.M. Data-Analytics-Driven Selection of Die Material in Multi-Material Co-Extrusion of Ti-Mg Alloys. *Mathematics* **2024**, *12*, 813. <https://doi.org/10.3390/math12060813>

Academic Editors: Yu-Wang Chen, Mi Zhou and Tao Wen

Received: 1 February 2024

Revised: 1 March 2024

Accepted: 5 March 2024

Published: 10 March 2024



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Keywords: data analytics; methodologies; multi-material; co-extrusion; FEM; machine learning; SVR; MCDM

MSC: 90-11; 90B25; 90B30; 90B50

1. Introduction

In recent years and with the rise of Industry 4.0, simulation and data analytics methodologies have become more relevant due to their capacity for predicting results and being more sustainable compared to traditional approaches. There is an increasing need to develop lighter materials in the aerospace and automotive industries to improve fuel efficiency, reduce environmental impacts, and increase the payloads to be carried out. Multi-material forming has become a solution because of its capacity to reduce weight by joining dissimilar materials and customize the mechanical properties of the final part to fulfill in-service requirements.

Aluminum alloys and carbon-fiber-reinforced composites are widely used in the aerospace industry. However, magnesium alloys have a lower density and a good specific strength [1]; therefore, they could be a good alternative to reduce weight if it were not for their poor corrosion resistance. Because of this, a multi-material component made of magnesium alloys combined with titanium alloys, which has excellent mechanical and physical–chemical properties, together with a very good strength-to-weight ratio

and superior corrosion resistance [2], could be the solution to manufacture lighter parts with good mechanical properties and without corrosion problems and thus contribute to reducing the weight of aircraft.

Multi-material forming involves a co-extrusion process to obtain bimetallic billets composed of a cylindrical sleeve and a core made of different materials. Some application cases of multi-material forming processes with two alloys are the studies performed by Fernández et al. [3,4], who analyzed the effects of different co-extrusion process parameters via finite element analysis (FEM) simulations, while using analysis of variance (ANOVA) to determine the most relevant parameters; these authors also investigated the effect of the selection of die material on the co-extrusion process of bimetallic cylindrical billets made of a magnesium alloy core and a titanium alloy sleeve. Other interesting contributions are the studies performed by Negendanka et al. [5], who carried out a study examining diffusion layer formation under different die angle values in a Mg-core and Al-sleeve billet, and by Gall et al. [6], who studied the co-extrusion of bimetallic Al-Mg billets into hollow profiles by means of a finite element method (FEM) simulation together with experiments.

On the other hand, machine learning (ML) [7] has been gaining traction in industry as the preferred method to forecast results and anticipate problems [8] by means of algorithms based on statistical methods to detect patterns from data. The support vector machine (SVM) is one of the most popular supervised learning methods within ML. It was introduced by Vladimir Vapnik [9] in 1995, and its main applications are in classification and regression analysis. For the latter, the implementation of a support vector regression (SVR) module within the SVM to estimate discrete values and thus predict future results is especially interesting. Some examples of SVR applications in industry are the prediction of the laser cutting process cost for AISI316L stainless steel [10], prediction of the cutting force and temperature in bone drilling [11], and prediction of the drilling force for drilling an internal hole in a carbon-fiber-reinforced polymer (CFRP) [12]. When applied to wear prediction, the research performed by Benkedjouh et al. can be highlighted [13].

Apart from ML, there are other approaches that allow decisions to be made in situations where there are several requirements to fulfill in a complex environment involving a large number of variables. Multi-criteria decision-making (MCDM) methods based on multi-objective optimization have been applied to find a compromise solution to this problem. The first MCDM method was applied by Pareto in 1896 [14], with his famous 80/20 principle. Another example is the study by Saaty in 1977 [15], in which multi-criteria models were used to solve problems with conflicting goals. Several MCDM methods have been developed and applied to support decision making in different areas, such as manufacturing process selection [16], supply chain management contract selection [17], and material selection [18]. In this research area, a combination of VIKOR [19] with entropy weighting methods [20,21] has been chosen as an MCDM methodology to establish optimum die material selection. In the study performed by Fernández et al. [22], different ARAS [23], TOPSIS [24], VIKOR, and COPRAS [25] MCDM methods were compared, in conjunction with the AHP [26], standard deviation [27], and entropy weighting methods.

This study developed two methodologies—one based on SVR and the other applying an entropy weighting method together with MCDM VIKOR—for the material selection of the die in a multi-material co-extrusion process to obtain bimetallic billets made of Ti6Al4V-AZ31B. Both methodologies and their results were compared to establish which one gives better results for the proposed problem.

2. Materials and Methods

2.1. Materials, Geometrical Dimensions, and Process Parameters

In this study, a bimetallic billet made of a Ti6Al4V titanium alloy sleeve and an AZ31B magnesium alloy core during the co-extrusion process was analyzed.

Figure 1 shows the co-extrusion setup with the process parameters and initial dimensions.

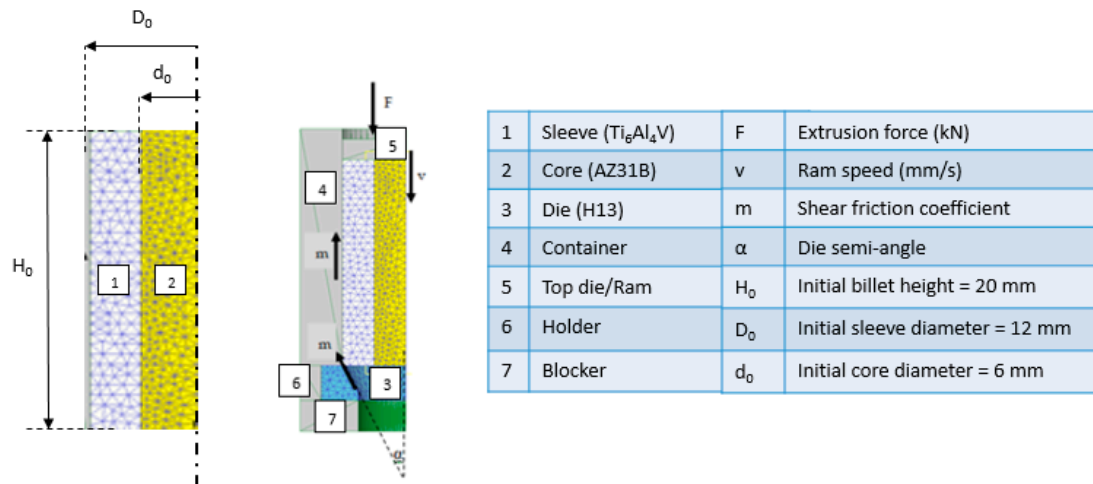


Figure 1. Co-extrusion setup with process parameters and initial billet dimensions.

The main physical and mechanical properties of Ti6Al4V and AZ31B are shown in Table 1.

Table 1. Physical and mechanical properties of the titanium alloy Ti6Al4V and the magnesium alloy AZ31B [28,29].

Property	Ti6Al4V	AZ31B
Density (g/cm ³)	4.46	1.74
Tensile strength (MPa)	895	260
Yield strength (MPa)	828	200
Elastic modulus (GPa)	110	44.80
Poisson's ratio	0.31	0.35

The chemical compositions of Ti6Al4V and AZ31B are listed in Tables 2 and 3, respectively.

Table 2. Chemical composition of titanium alloy Ti6Al4V [28].

Ti (wt.%)	Al (wt.%)	V (wt.%)	Fe (wt.%)	C (wt.%)	O (wt.%)	N (wt.%)	H (wt.%)
Bal.	5.5–6.5	3.5–4.5	0.25	0.08	0.13	0.040	0.012

Table 3. Chemical composition of magnesium alloy AZ31B [29].

Mg (wt.%)	Al (wt.%)	Zn (wt.%)	Mn (wt.%)	Si (wt.%)	Cu (wt.%)	Ca (wt.%)	Fe (wt.%)	Ni (wt.%)
97	2.5–3.5	0.6–1.4	0.20	0.1	0.05	0.04	0.005	0.005

The material candidates for the die were extracted from Daniel et al. [4], whose chemical composition and physical and mechanical properties data are shown in Tables 4 and 5, respectively.

Table 4. Chemical composition of die steels [30–34].

Material	C (wt.%)	Mn (wt.%)	Si (wt.%)	Cr (wt.%)	Mo (wt.%)	Ni (wt.%)
AISI316	0.08	2	0.75	16–18	2–3	10–14
H13	0.32–0.45	0.2–0.5	0.8–1.20	4.75–5.50	1.10–1.75	0.30 max
25CrMo4	0.22–0.29	0.60–0.90	0.10–0.40	0.90–1.20	0.15–0.30	-
AISI52100	0.1	0.45	0.26	1.51	0.06	3.39
AISI3310	0.99	0.39	0.16	1.4	-	1.4

Table 5. Physical and mechanical properties of die steels [35].

Property	AISI316	H13	25CrMo4	AISI52100	AISI3310
Density (g/cm ³)	8.03	7.78	7.85	7.83	7.81
Tensile strength (MPa)	550	1990	670	992	1866
Yield strength (MPa)	240	1650	435	579	1800
Elastic modulus (GPa)	210	210	205	200	210
Poisson's ratio	0.3	0.3	0.3	0.3	0.3

The extrusion process parameters evaluated in this research are the following:

- Process parameters: ram speed (mm/s) and temperature (°C).
- Tooling parameters: die semi-angle (°), shear friction factor, and extrusion ratio (A_0/A_f).
- Geometric parameters: shape factor (H_0/D_0) and diameter ratio (D_0/d_0).

In these parameters, A_0 and A_f are the initial area and the final area of the cross-section of the billet, respectively; D_0 and d_0 are the initial external diameter and the internal diameter of the sleeve, respectively; and H_0 is the initial billet height.

2.2. Finite Element Modeling and Simulation Preparation

The commercial software DEFORM3D© (v11.2) [36] was used to perform the finite element simulations.

All parts were meshed with 7000 tetrahedral elements, and due to the axial symmetry of the process, only one-quarter of the problem was modeled to reduce the computation time and to avoid the generation of heavy database files.

The contact condition among the objects of the simulation was defined as follows: rigid and elastic objects were considered the “masters” (those that deform) and plastic objects were considered “slaves” (those that are deformed). In the case of the interaction between the sleeve and the core, where both objects are plastic, the titanium alloy was defined as the “master” and the magnesium alloy was defined as the “slave”. All materials were assumed to be isotropic throughout the process.

The heat transfer coefficient between the sleeve and the core and between the sleeve and the die was set to 11 N/(s·mm·°C), while that between the extrusion tooling elements and the die was set to 5 N/(s·mm·°C). All objects of the simulation had a 0.02 N/(s·mm·°C) heat transfer coefficient with the air.

The exponential model developed by Wen-juan et al. [37] was used to define the behavior of AZ31B, while the Johnson–Cook constitutive equations [38] were used for the definition of the stress–strain curves of Ti6Al4V.

2.2.1. Tool Wear Model

Archard's wear model was used to calculate the wear produced on the surface of the die [39–41]. This model is based on Equation (1):

$$W = \int K \times \frac{p^a \cdot v^b}{H^c} \times dt \quad (1)$$

where K is the wear coefficient; P is the interface pressure; v is the sliding velocity between the die and the billet; H is the hardness; and a , b , and c are experimentally calibrated coefficients.

The commonly taken value for a and b is 1, while for c , the value is 2 in the case of steel alloys. The equation for calculating the wear coefficient is as follows:

$$K = 2 \times 10^{-5}.$$

Taking into account Equation (1), the parameters used to evaluate the wear were the ram speed and friction, as they can influence the sliding velocity, as well as temperature, because it has a direct influence on the stress–strain curves.

2.2.2. FEM Validation

The FEM was validated by using the semi-empirical model proposed by Johnson, which was applied by García et al. [42]. This model is typically used as a reference in extrusion processes to establish an upper limit of the extrusion force. To apply Johnson's model, it is necessary to obtain the average yield stress of each component in accordance with its volume fraction, as described by Gisbert et al. [43]. The force obtained by the FEM was in good agreement with the semi-empirical model results, with this force being the upper limit of the required forces, as expected. Thus, the FE model can be considered to have been validated.

2.3. Support Vector Regression

The SVM works by finding a hyperplane in a high-dimensional space that best separates data into different classes. It aims to maximize the margin (the distance between the hyperplane and the nearest data points of each class) while minimizing classification errors. The SVM can handle both linear and non-linear classification problems by using various kernel functions. Unlike the SVM used for classification tasks, SVR seeks to find a hyperplane that best fits the data points in a continuous space.

SVR [44] offers the flexibility to define how much error is acceptable in a model and is used to find an appropriate line (or a hyperplane in higher dimensions) to fit the data. Therefore, the goal of SVR is to find a function that approximates the relationship between the input variables and a continuous target variable while minimizing the prediction error.

Another advantage of SVR over linear or logistic regression is the possibility of using different kernel functions, such as polynomial or radial functions, which allows the transformation of data into a higher dimensional space, thereby making it suitable for non-linear problems.

As mentioned before, the idea is to minimize the result of Equation (2), while taking into account the constraints of Equations (3)–(5), as it is shown in Figure 2:

$$\frac{1}{2} \|w\|^2 + C \sum_{i=1}^N (\xi_i + \xi_i^*) \quad (2)$$

$$y_i - wx_i - b \leq \varepsilon + \xi_i \quad (3)$$

$$wx_i + b - y_i \leq \varepsilon + \xi_i^* \quad (4)$$

$$\xi_i, \xi_i^* \geq 0 \quad (5)$$

where ε is the margin of error; ξ is the deviation from ε , which is also called the tolerance margin; and w is the classification vector. C is known as the regularized parameter.

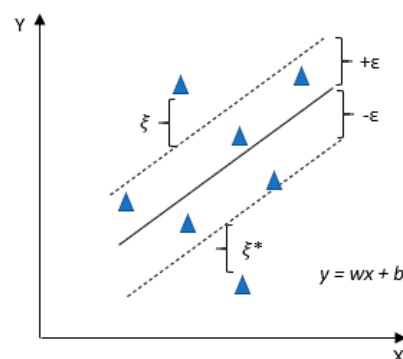


Figure 2. Two-dimensional hyperplane representation.

The prediction error can be calculated in different ways. One of the most representative is by using the determination factor (R^2), which shows the quality of the correlation between the real measured data and the value predicted by Equation (6). A more precise

correlation will be obtained when the value of the determination factor is nearer to 100%. The calculation is as follows:

$$R^2 = \frac{[\sum_{i=1}^n (\theta_i - \theta_i^{mean})(\hat{\theta}_i - \hat{\theta}_i^{mean})]^2}{[\sum_{i=1}^n (\theta_i - \theta_i^{mean})][\sum_{i=1}^n (\hat{\theta}_i - \hat{\theta}_i^{mean})]} \quad (6)$$

where θ_i is the measurement data, $\hat{\theta}_i$ is the predicted magnitude in accordance with SVR, θ_i^{mean} is the mean of the measurement data, and $\hat{\theta}_i^{mean}$ is the mean of the prediction.

2.4. Entropy Method

The entropy method [21,22] is classified within the category of objective weighting methods, and it is applicable when the data of a decision matrix are known. Entropy is a measure of randomness and disorder in the universe.

Starting with the decision matrix D , the projected outcomes p_{ij} are calculated by means of Equation (7):

$$D = \begin{bmatrix} x_{11} & \cdots & x_{1n} \\ \vdots & \ddots & \vdots \\ x_{m1} & \cdots & x_{mn} \end{bmatrix}$$

$$p_{ij} = \frac{x_{ij}}{\sum_{i=0}^m x_{ij}} \quad (7)$$

where n is the number of criteria and m is the number of alternatives.

The entropy measure of the projected outcomes is obtained as shown in Equation (8):

$$E_j = -k * \sum_{i=1}^m p_{ij} * \ln(p_{ij}) \quad (8)$$

with $k = 1/\ln(m)$.

The objective weight-based definition is calculated according to Equation (9):

$$w_j = \frac{1 - E_j}{\sum_{j=1}^n (1 - E_j)} \quad (9)$$

2.5. VIKOR Method

VIKOR [45,46] stands for VIseKriterijumska Optimizacija I Kompromisno Resenje, which means multi-criteria optimization and compromise solution.

This methodology is based on the concept that the compromise solution is the one that is at the minimum distance from the ideal solution while, at the same time, at the maximum distance from the anti-ideal solution.

One big difference from the SVM is that VIKOR does not require the calculation of the error because there is no prediction and therefore there is nothing to compare. Instead, VIKOR requires a validation step before declaring the compromise solution that is feasible by fulfilling the “acceptable advantage” and “acceptable stability in decision making” conditions.

Other advantages of using the VIKOR method are the following:

- The ability to immediately recognize the proper alternative;
- A decrease in the number of pairwise comparisons required.

After the criteria to be evaluated are defined, a decision matrix (D) is built:

$$D = \begin{bmatrix} x_{11} & \cdots & x_{1n} \\ \vdots & \ddots & \vdots \\ x_{m1} & \cdots & x_{mn} \end{bmatrix}$$

At this point, the values of the best f_b^* and the worst f_b^- for each criterion rating of the decision matrix are obtained as follows:

$f_b^* = \max(x_{ib})$ and $f_b^- = \min(x_{ib})$, when the objective is to maximize the criteria.

$f_b^* = \min(x_{ib})$ and $f_b^- = \max(x_{ib})$, when the objective is to minimize the criteria.

Where $b = 1 \dots m$, with m being the number of criteria that is taken into account, and $i = 1 \dots n$, with n being the number of alternatives.

The Utility measure (S_j) and the Regret measure (R_j) are calculated according to Equations (10) and (11):

$$S_j = \sum_{b=1}^m W_b * \left[\frac{f_b^* - f_{ij}}{f_b^* - f_b^-} \right] \quad (10)$$

$$R_j = \max \left[W_b * \left[\frac{f_b^* - f_{ij}}{f_b^* - f_b^-} \right] \right] \quad (11)$$

where W_b are the weight values obtained in this study after applying the entropy weighting methods explained above.

The index Q can be obtained by using Equation (12):

$$Q_a = v * \frac{S_j - S^*}{S^- - S^*} + (1 - v) * \frac{R_j - R^*}{R^- - R^*} \quad (12)$$

where

$$S^- = \max(S_j)$$

$$S^* = \min(S_j)$$

$$R^- = \max(R_j)$$

$$R^* = \min(R_j)$$

In this equation, v is a parameter that represents the type of voting used during the process ($v > 0.5$ means “vote by majority rule”, $v = 0.5$ means “vote by consensus”, and $v < 0.5$ means “with vote”).

The lowest Q_a value indicates the best alternative solution, and it can be recommended if the following conditions are satisfied:

The “acceptable advantage” condition means that $Q(a'') - Q(a') \geq DQ$, with a'' being the alternative in the second position in the ranking list by Q_a and a' being the first one. DQ is defined according to Equation (13):

$$DQ = \frac{1}{(n-1)} \quad (13)$$

where n is the number of alternatives.

Finally, the “acceptable stability in decision making” condition implies that the a' alternative must also be the best ranked for S_j and/or R_j . If one of the conditions is not fulfilled, then a set of compromise solutions is proposed.

2.6. Methodology

Two different methodologies were proposed for the selection of the optimum die material in order to obtain the minimum extrusion force and die wear. The methodology steps are shown in the flowchart in Figure 3.

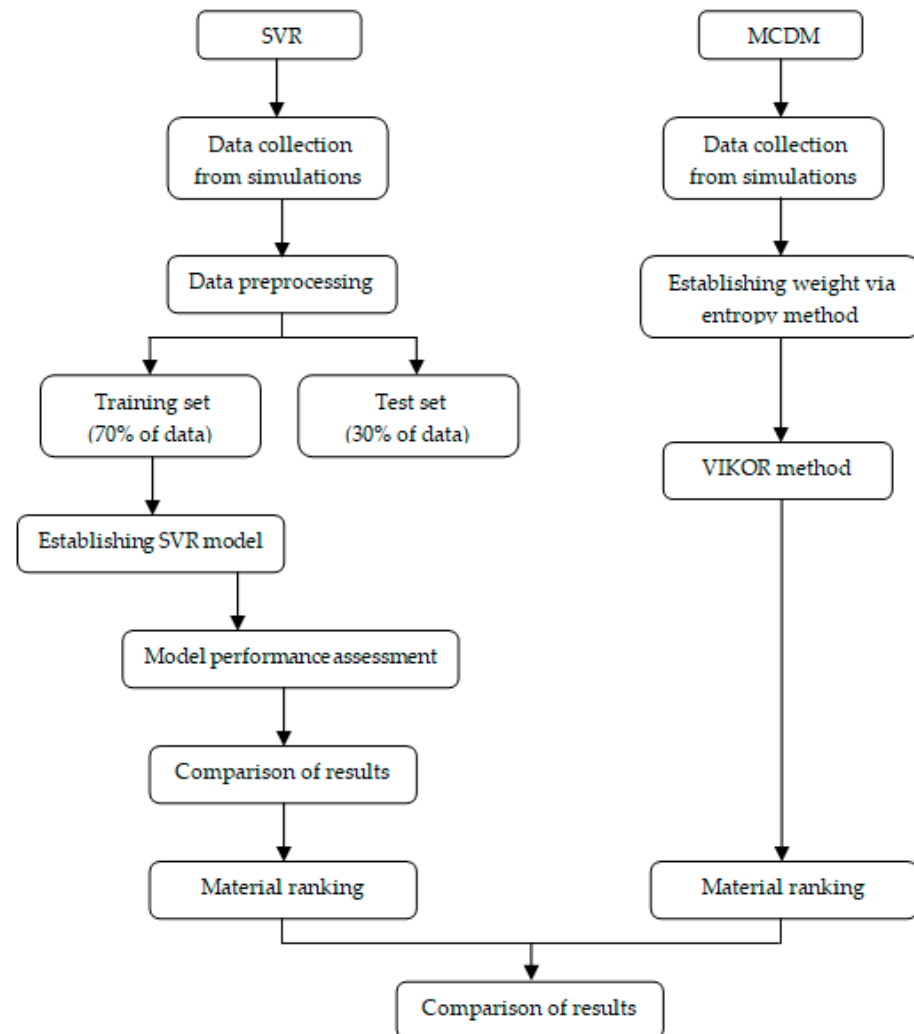


Figure 3. Methodology flowchart.

The criteria for the final result comparison are as follows:

- Simplicity;
- Amount of data obtained from the simulations;
- Time consumption.

The prediction was carried out using the Python software [47].

3. Results

In this study, a set of simulations for a multi-material co-extrusion process were performed by using the commercial software DEFORM3D® (v11.2), followed by the application of two different methodologies, to choose the best die material to obtain the minimum extrusion force and the minimum wear during the process. For the list of simulations carried out in the present work, see Table A1 in Appendix A.

3.1. SVR Methodology

As explained above, a dataset was obtained for the parameters listed in Table 6, and for each material and each parameter to be predicted, several dataframes were obtained by using the “pandas” together with the “sklearn” libraries.

Table 6. Ranking of process parameters' influence on extrusion force.

Material	Parameters
AISI316	Extrusion ratio, friction, ram speed, core diameter, billet height, die semi-angle, and temperature.
H13	Friction, extrusion ratio, core diameter, billet height, die semi-angle, ram speed, and temperature.
25CrMo4	Friction, ram speed, billet height, core diameter, die semi-angle, temperature, and extrusion ratio.
AISI52100	Friction, core diameter, die semi-angle, extrusion ratio, billet height, ram speed, and temperature.
AISI3310	Ram speed, core diameter, friction, extrusion ratio, die semi-angle, billet height, and temperature.

Using the “RFE” module for Regression Feature Selection from “sklearn.feature_selection”, together with the “SVR” module from “sklearn.svm”, the process parameters were ranked according to their influence on the extrusion force, as shown in Table 6.

Taking into account these results, it can be concluded that friction is the most important process parameter, while temperature is the least important one. This conclusion is in good agreement with the findings reported by Fernández et al. [3,4], who performed a deeper analysis of the influence of each process parameter on the extrusion force.

As there is no clear pattern regarding the influence of the process parameters and their influence is clearly dependent on the die material, all the parameters were implemented in the prediction model.

For the prediction model of the extrusion force, the dataframes for each material were split into two groups—one for training and one for testing—using the “train_test_split” function from the “sklearn.model_selection” module, with the test size being 0.3.

After applying the “LinearRegression” function from “sklearn.linear_model” to build the prediction model using the training data, the model was evaluated using the test data. The determination factor (R^2) for each material is shown in Table 7.

Table 7. Determination factor (R^2) for the extrusion force in the linear regression model.

Material	R^2
AISI316	0.91461
H13	0.92245
25CrMo4	0.70708
AISI52100	0.86922
AISI3310	0.91966

Figures 4–8 show the comparisons between the simulation values and the prediction values.

By using the prediction model for each die material, a larger number of results for the extrusion force could be compared without the need to perform more simulations. Table 8 shows the ranking of the die materials as a function of the times that their prediction value for the extrusion force was the lowest.

If the minimum extrusion force was the only requirement for the selection of die materials, AISI3310 would be the chosen material, followed by AISI316 and H13, which shared the second position in the ranking.

The SVR methodology was then applied for wear prediction with the following modification:

Due to the variation in the results, it was not possible to apply a linear regression model and so a polynomial one had to be used. To produce a polynomial regression model, it was necessary to import the “PolynomialFeatures” module from the “sklearn.preprocessing” library to generate a new feature matrix consisting of all the polynomial combinations of the features with a degree less than or equal to the specified degree (in this case, a two-degree polynomial function was used).

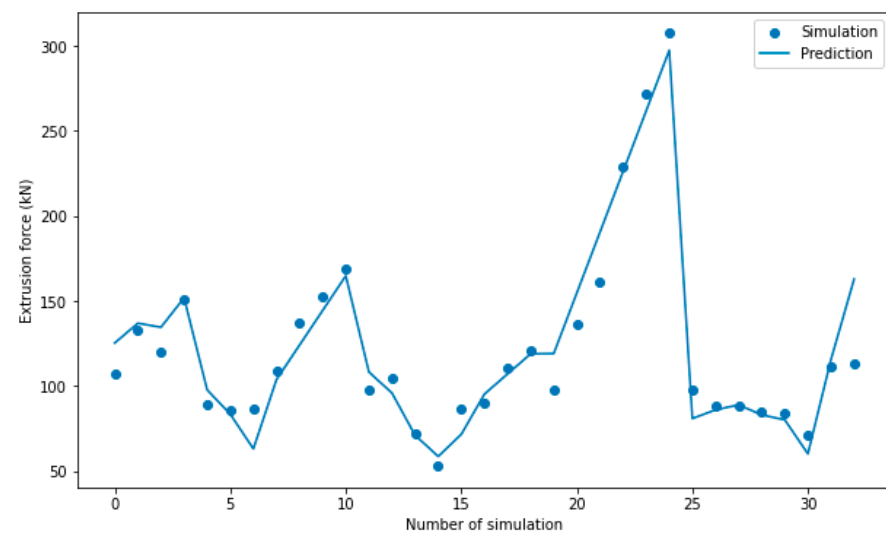


Figure 4. Comparison of AISI316 extrusion force prediction and simulation values.

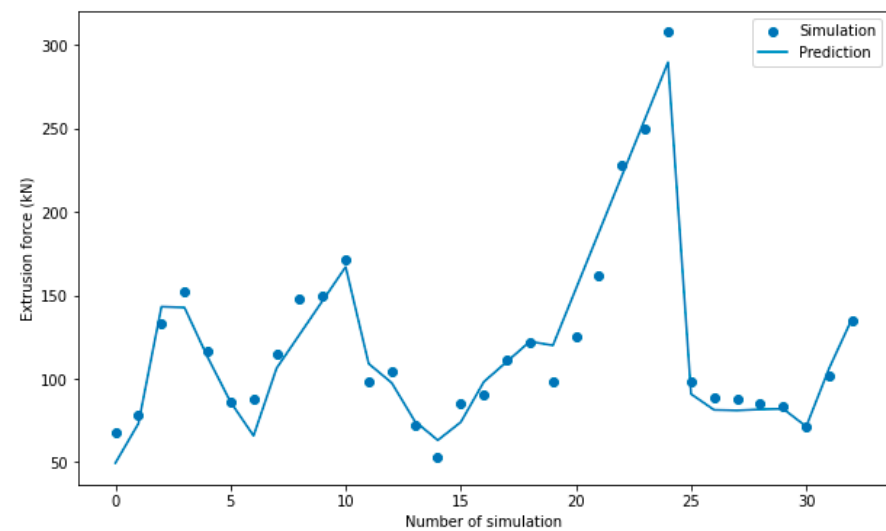


Figure 5. Comparison of H13 extrusion force prediction and simulation values.

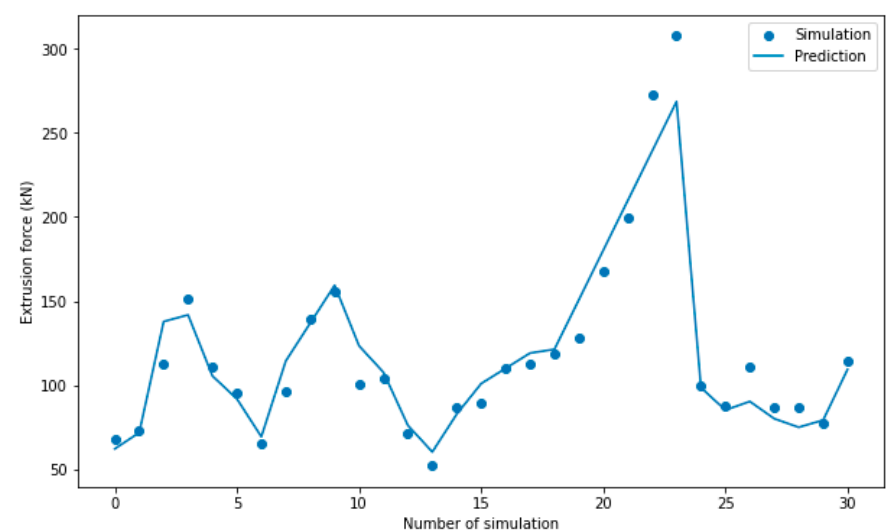


Figure 6. Comparison of 25CrMo4 extrusion force prediction and simulation values.

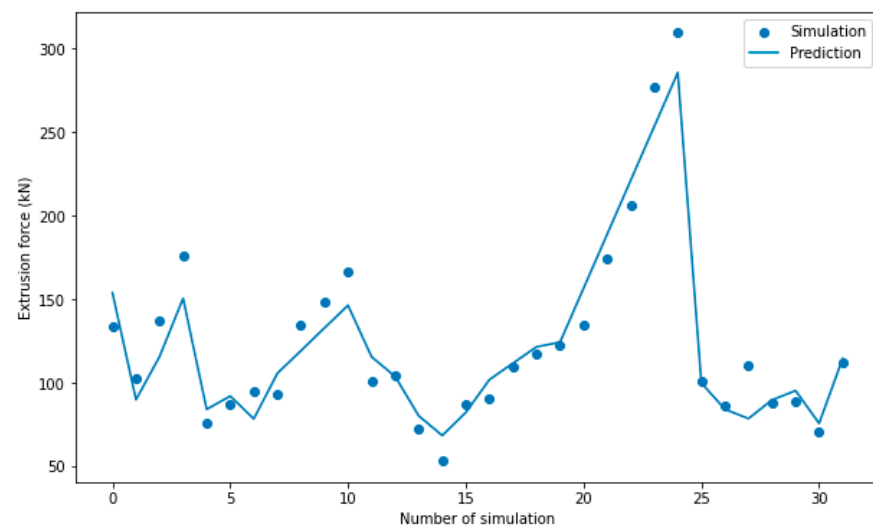


Figure 7. Comparison of AISI52100 extrusion force prediction and simulation values.

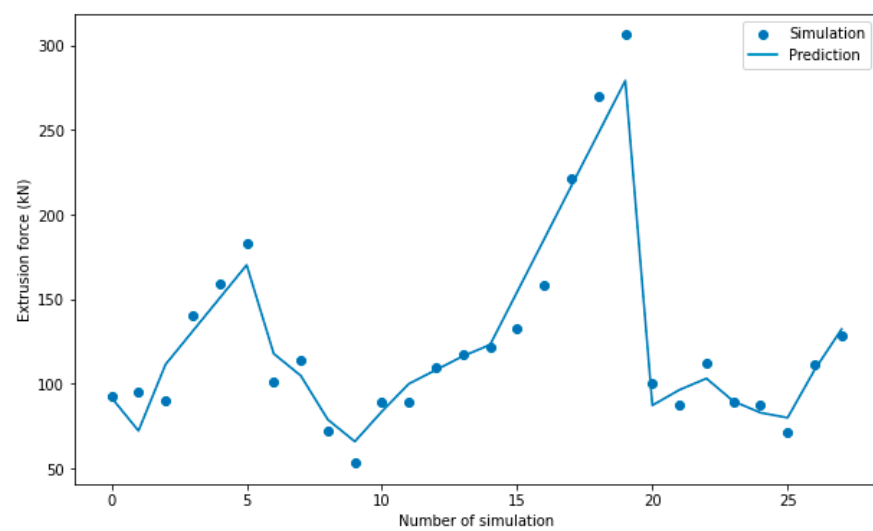


Figure 8. Comparison of AISI3310 extrusion force prediction and simulation values.

Table 8. Ranking of die materials as a function of the times that the lowest extrusion force was produced.

Material	Ranking
AISI316	2
H13	2
25CrMo4	5
AISI52100	3
AISI3310	1

Tables 9 and 10 show the ranking of the process parameters and the determination factor (R^2) for the wear model.

Table 9. Ranking of process parameters for die wear.

Material	Parameters
AISI316	Friction, ram speed, and temperature.
H13	Friction, ram speed, and temperature.
25CrMo4	Temperature, friction, and ram speed.
AISI52100	Friction, temperature, and ram speed.
AISI3310	Temperature, ram speed, and friction.

Table 10. Determination factor (R^2) for the extrusion force in the polynomial regression model.

Material	R^2
AISI316	0.75695
H13	0.74873
25CrMo4	0.63223
AISI52100	0.80571
AISI3310	0.65881

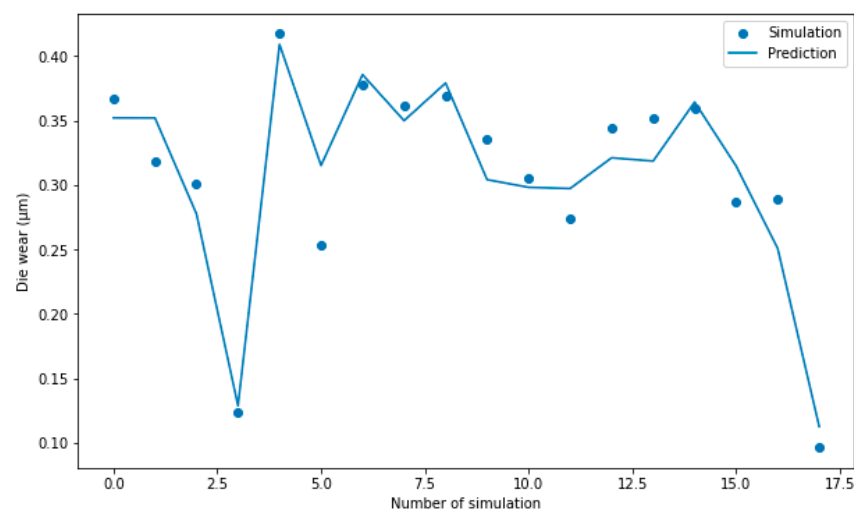
The conclusion is in good agreement with the findings reported by Fernández et al. [4], who performed a deeper analysis of the influence of each process parameter on the die wear.

The prediction model for the die wear was not as accurate as the one for the extrusion force. This could be due to the fact that the number of simulations performed to obtain the wear distribution was lower than for that for the extrusion force because Archard's wear model only takes into account temperature, friction, and ram speed, as mentioned in Section 2.2.1.

Figures 9–13 show the comparisons between the simulation values and the prediction values.

Table 11 shows the ranking of the die materials as a function of the times that their prediction value for the die wear was the lowest value.

Finally, a crosscheck between Tables 8 and 11 was performed to rank the die materials and find the one that best fulfills the requirements of minimum extrusion force and minimum die wear, and the results are shown in Table 12.

**Figure 9.** Comparison of AISI316 die wear prediction and simulation values.

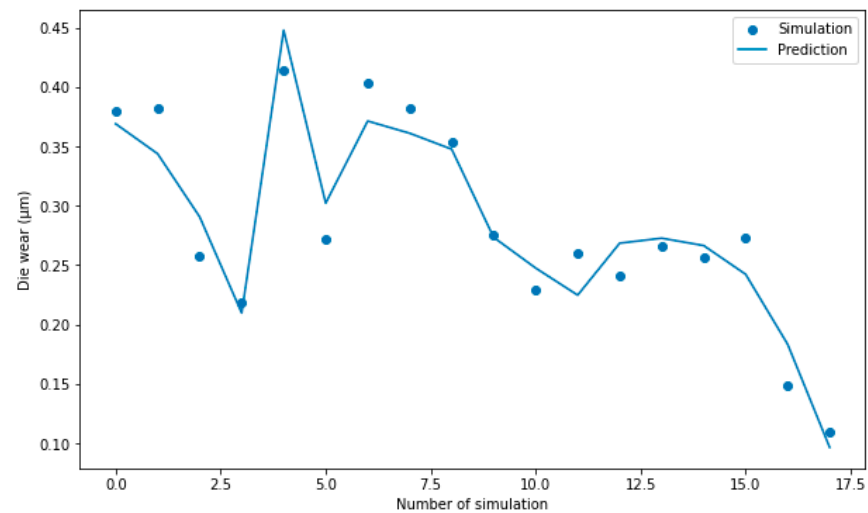


Figure 10. Comparison of H13 die wear prediction and simulation values.

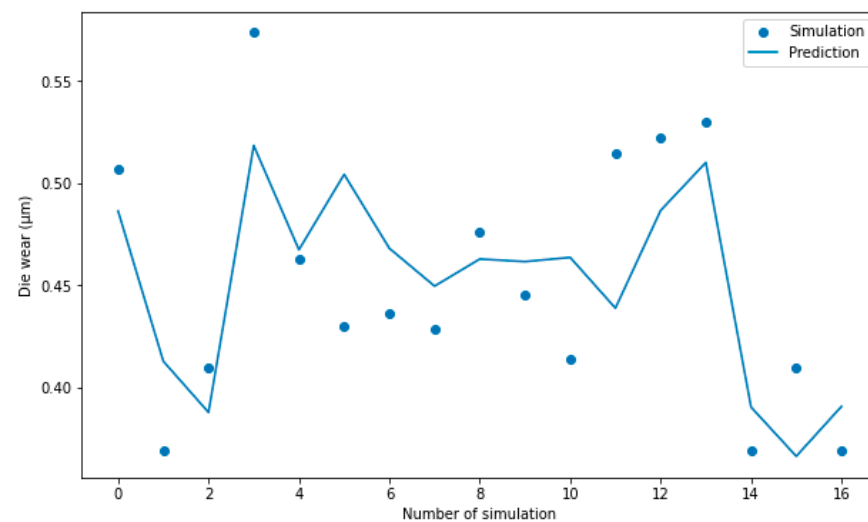


Figure 11. Comparison of 25CrMo4 die wear prediction and simulation values.

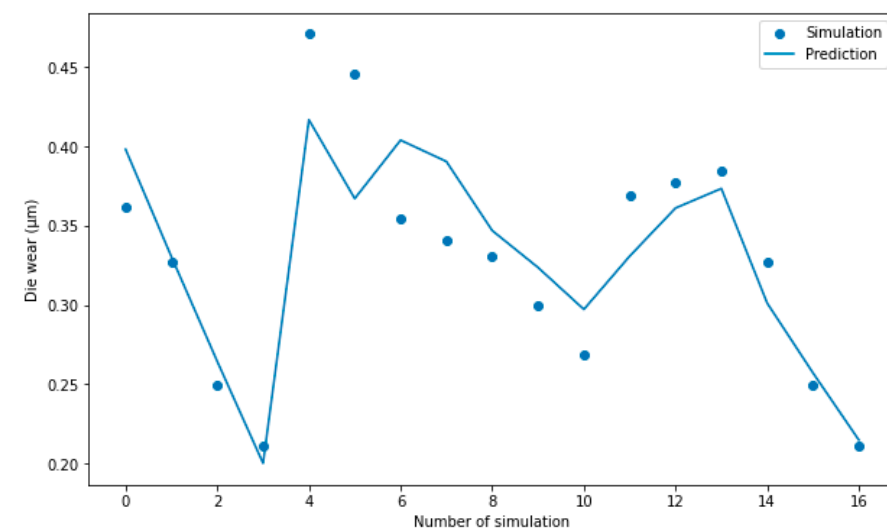


Figure 12. Comparison of AISI52100 die wear prediction and simulation values.

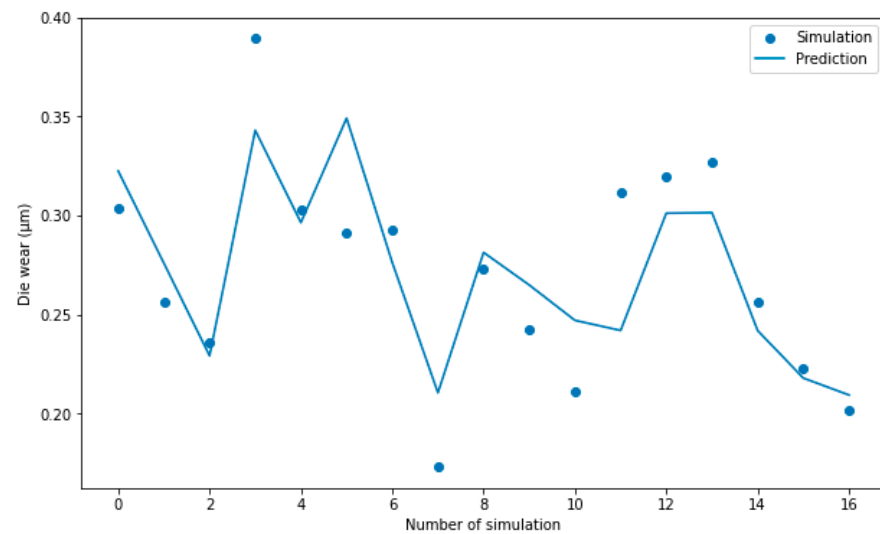


Figure 13. Comparison of AISI3310 die wear prediction and simulation values.

Table 11. Ranking of die materials as a function of the times that the minimum wear in the die was produced.

Material	Ranking
AISI316	2
H13	3
25CrMo4	N/A
AISI52100	N/A
AISI3310	1

Table 12. Ranking of die materials as a function of fulfilling the requirements of minimum extrusion force and minimum die wear.

Material	Ranking
AISI316	2
H13	3
25CrMo4	5
AISI52100	4
AISI3310	1

In both rankings, AISI3310 is the best choice to reduce the extrusion force and die wear. AISI316 and H13 have the same ranking position with regard to the extrusion force but not with regard to the die wear, and the reason is because, in the final ranking, AISI316 is ranked better than H13. The die materials that can be rejected as feasible options are AISI52100 and 25CrMo4.

3.2. MCDM Methodology

As explained in Section 3, a dataset was obtained from the simulations listed in Table A1. In the MCDM methodology, weights were calculated by means of the entropy method and, afterwards, the VIKOR method was applied to classify the different materials based on the criteria rating values.

For the entropy method, the decision matrix D was obtained based on the results for each simulation listed in Table A1 regarding the extrusion force and die wear.

86.019	136.529	228.511	307.324	97.898	88.621	88.040	84.512	83.759	0.367	0.318	0.302	0.124	0.417	0.253	0.378	0.361	0.370
86.153	124.955	227.644	307.783	98.095	88.706	88.140	84.799	83.850	0.379	0.382	0.257	0.219	0.414	0.271	0.404	0.381	0.354
95.608	128.196	199.579	307.520	100.048	87.505	111.049	87.149	86.484	0.361	0.327	0.250	0.211	0.471	0.445	0.354	0.341	0.345
87.120	134.333	206.128	309.397	100.654	86.238	110.203	88.307	88.602	0.507	0.369	0.410	0.452	0.573	0.462	0.430	0.436	0.428
92.531	132.578	221.350	306.154	100.708	87.422	112.221	89.296	87.588	0.304	0.257	0.281	0.236	0.389	0.303	0.291	0.292	0.173

From D , the normalized matrix was obtained by means of Equation (7):

0.1923	0.2079	0.2110	0.1998	0.1968	0.2021	0.1727	0.1947	0.1947	0.1913	0.1925	0.2011	0.1000	0.1842	0.1459	0.2035	0.1993	0.2214
0.1926	0.1903	0.2102	0.2001	0.1972	0.2023	0.1729	0.1954	0.1949	0.1977	0.2311	0.1717	0.1764	0.1829	0.1563	0.2175	0.2105	0.2118
0.2137	0.1952	0.1842	0.1999	0.2011	0.1996	0.2179	0.2008	0.2010	0.1884	0.1978	0.1667	0.1701	0.2078	0.2568	0.1908	0.1881	0.2067
0.1947	0.2046	0.1903	0.2011	0.2024	0.1967	0.2162	0.2034	0.2059	0.2642	0.2233	0.2733	0.3639	0.2532	0.2665	0.2315	0.2407	0.2565
0.2068	0.2019	0.2043	0.1990	0.2025	0.1994	0.2202	0.2057	0.2036	0.1584	0.1553	0.1871	0.1897	0.1719	0.1744	0.1567	0.1614	0.1037

Then, the entropy array (E_j) was calculated by applying Equation (8):

$E_j = [0.999418867 \ 0.999681506 \ 0.999085371 \ 0.999996437 \ 0.999952024 \ 0.999966715 \ 0.996102816 \ 0.999852667$
 $0.999839187 \ 0.990960752 \ 0.99430503 \ 0.989129167 \ 0.944884512 \ 0.993690857 \ 0.979914551 \ 0.994771731 \ 0.99470602$
 $0.976990905]$

The weights are presented in Table 13.

Table 13. Entropy method weights.

W. 1(%)	W. 2(%)	W. 3(%)	W. 4(%)	W. 5(%)	W. 6(%)	W. 7(%)	W. 8(%)	W. 9(%)	W. 10(%)	W. 11(%)	W. 12(%)	W. 13(%)	W. 14(%)	W. 15(%)	W. 16(%)	W. 17(%)	W. 18(%)
0.40	0.22	0.62	0	0.03	0.02	2.66	0.10	0.11	6.16	3.88	7.41	37.56	4.30	13.69	3.56	3.61	15.68

Using the VIKOR method, the best f_b^* and worst f_b^- values for each criterion were obtained directly from the decision matrix D .

	86.019	136.529	228.511	307.324	97.898	88.621	88.040	84.512	83.759	0.367	0.318	0.302	0.124	0.417	0.253	0.378	0.361	0.370
	86.153	124.955	227.644	307.783	98.095	88.706	88.140	84.799	83.850	0.379	0.382	0.257	0.219	0.414	0.271	0.404	0.381	0.354
	95.608	128.196	199.579	307.520	100.048	87.505	111.049	87.149	86.484	0.361	0.327	0.250	0.211	0.471	0.445	0.354	0.341	0.345
	87.120	134.333	206.128	309.397	100.654	86.238	110.203	88.307	88.602	0.507	0.369	0.410	0.452	0.573	0.462	0.430	0.436	0.428
	92.531	132.578	221.350	306.154	100.708	87.422	112.221	89.296	87.588	0.304	0.257	0.281	0.236	0.389	0.303	0.291	0.292	0.173
	447.431	656.591	1083.212	1538.178	497.402	438.492	509.653	434.063	430.283	1.918	1.652	1.499	1.242	2.265	1.735	1.856	1.812	1.669
f_i^*	86.019	124.955	199.579	306.154	97.898	86.238	88.040	84.512	83.759	0.304	0.257	0.250	0.124	0.389	0.253	0.291	0.292	0.173
f_i^-	95.608	136.529	228.511	309.397	100.708	88.706	112.221	89.296	88.602	0.507	0.382	0.410	0.452	0.573	0.462	0.430	0.436	0.428

The Utility measure (S_j) and the Regret measure (R_j) were then obtained:

	S_j	R_i
	0.23758265	0.12075866
	0.36022622	0.11092085
	0.44932592	0.1258087
	0.98459134	0.37557176
	0.21183731	0.12764252
S^*	0.21183731	R^* 0.11092085
S^-	0.98459134	R^- 0.37557176

Using the values S^* , S^- , R^* , and R^- , together with the assumption of vote by consensus ($v = 0.5$), the index Q was calculated:

	Q_i
AISI3310	0.03524455
H13	0.09601303
AISI52100	0.18179111
25CrMo4	1
AISI3310	0.03159193

Using the VIKOR method, the index Q was ranked from the lowest to the highest value. The best material to obtain the minimum extrusion force and the minimum die wear is AISI3310. But before recommending this material as the best compromise solution, the conditions of “acceptable advantages” and “acceptable stability in decision making” have to be fulfilled.

In this case, $DQ = 0.25$ according to Equation (13).

$$\begin{aligned}
 Q(2) - Q(1) &= 0.0365261 \\
 Q(3) - Q(1) &= 0.0644211 \\
 Q(4) - Q(1) &= 0.15019917 \\
 Q(5) - Q(1) &= 0.96840807 > DQ \\
 Q(1) &= S^*
 \end{aligned}$$

As only the second condition was fulfilled, a set of compromise solutions was ranked, and the results are presented in Table 14.

Table 14. VIKOR ranking of die materials.

Material	Ranking
AISI316	2
H13	3
25CrMo4	4
AISI52100	5
AISI3310	1

4. Discussion

In this paper, two methodologies are proposed to choose the best material for the die in a multi-material coextrusion process, taking into account that the process has to fulfill the requirements of minimum extrusion force and minimum die wear.

The first methodology proposed is an SVR methodology based on the SVM. The main advantage of this methodology is that the prediction model obtained during the process allows engineers to predict the outcomes when varying the process parameters. On the other hand, the disadvantages are the number of simulations needed to obtain a good prediction model and, depending on the results of those simulations, the complexity of obtaining the prediction model, which can be very high.

The second methodology is the MCDM methodology, which allows the selection of the best die material with a smaller number of simulations than the SVR methodology and without having to consider the accuracy or complexity of prediction models. Also, it is less time-consuming because the entropy and VIKOR methods can be applied directly to the data, and there is no need to have knowledge in programming languages like Python.

The results for the top three materials selected are the same regardless of the methodology applied. Therefore, if there is no need to apply other values to the parameters to obtain a prediction model to forecast the results, the recommended methodology for die material selection is the MCDM methodology due to its simplicity and the time saved in implementing it because there is no need for data preprocessing, nor there is a need for programming or statistical knowledge to interpret the results of the obtained model. In addition, MCDM is more scalable due to the fact that the SVM needs to re-evaluate the prediction model with new data to ensure the accuracy of the results while making certain that the prediction error is not increased.

Finally, for future research, it would be interesting to conduct a comparison among different machine learning methods to obtain a more robust prediction model not only for the wear but also for other parameters, such as the damage factor, mean stresses, and resultant microstructure.

Author Contributions: Conceptualization, D.F., Á.R.-P. and A.M.C.; methodology, D.F.; formal analysis, D.F., Á.R.-P. and A.M.C.; investigation, D.F., Á.R.-P. and A.M.C.; resources, Á.R.-P. and A.M.C.; writing—original draft preparation, D.F.; writing—review and editing, Á.R.-P. and A.M.C.; supervision, Á.R.-P. and A.M.C.; project administration, Á.R.-P. and A.M.C.; funding acquisition, Á.R.-P. and A.M.C. All authors have read and agreed to the published version of the manuscript.

Funding: This research was funded within the framework of the “Doctorate Program in Industrial Technologies” of the UNED and by the project 2021V/-TAJOV/006 (awarded as part of the call for UNED Research Projects for “Young Talents 2021”).

Data Availability Statement: The raw/processed data required to reproduce these findings cannot be shared at this time as the data also form part of an ongoing study.

Acknowledgments: We would like to extend our gratitude to the Research Group of the UNED “Industrial Production and Manufacturing Engineering (IPME)” and the Industrial Research Group “Advanced Failure Prognosis for Engineering Applications”.

Conflicts of Interest: The authors declare no conflicts of interest. The funders had no role in the design of the study; in the collection, analyses, or interpretation of data; in the writing of the manuscript; or in the decision to publish the results.

Appendix A

Table A1. List of simulations performed by using DEFORM3D® (v11.2).

Simulation	Material	Ram Speed (mm/s)	Core Diameter (mm)	Billet Height (H)	Temperature (°C)	Friction	Die Semi-Angle (°)	Extrusion Ratio
1	AISI316	2	5	20	200	0.2	30	1.78
2	AISI316	2	6	15	100	0.2	30	2.25
3	AISI316	2	7	25	100	0.3	30	1.44
4	AISI316	3	6	15	200	0.3	15	2.25
5	AISI316	3	7	15	300	0.2	45	1.44
6	AISI316	2	6	20	200	0.1	30	1.78
7	AISI316	2	6	20	200	0.1	15	1.78
8	AISI316	2	6	20	200	0.1	45	1.78
9	AISI316	2	6	20	200	0.1	60	1.78
10	AISI316	2	6	20	200	0.1	75	1.78
11	AISI316	2	6	20	200	0.1	90	1.78
12	AISI316	2	2	20	200	0.1	30	1.78
13	AISI316	2	4	20	200	0.1	30	1.78
14	AISI316	2	8	20	200	0.1	30	1.78
15	AISI316	2	10	20	200	0.1	30	1.78
16	AISI316	2	6	15	200	0.1	30	1.78
17	AISI316	2	6	25	200	0.1	30	1.78
18	AISI316	2	6	30	200	0.1	30	1.78
19	AISI316	2	6	35	200	0.1	30	1.78
20	AISI316	2	6	20	200	0.2	30	1.78
21	AISI316	2	6	20	200	0.3	30	1.78
22	AISI316	2	6	20	200	0.4	30	1.78
23	AISI316	2	6	20	200	0.5	30	1.78
24	AISI316	2	6	20	200	0.6	30	1.78
25	AISI316	2	6	20	200	0.7	30	1.78

Table A1. Cont.

Simulation	Material	Ram Speed (mm/s)	Core Diameter (mm)	Billet Height (H)	Temperature (° C)	Friction	Die Semi-Angle (°)	Extrusion Ratio
26	AISI316	2	6	20	100	0.1	30	1.78
27	AISI316	2	6	20	300	0.1	30	1.78
28	AISI316	1	6	20	300	0.1	30	1.78
29	AISI316	3	6	20	300	0.1	30	1.78
30	AISI316	4	6	20	300	0.1	30	1.78
31	AISI316	2	6	20	200	0.1	30	1.44
32	AISI316	2	6	20	200	0.1	30	2.25
33	AISI316	2	6	20	200	0.1	30	2.94
34	H13	2	5	15	100	0.1	15	1.44
35	H13	2	6	25	300	0.1	15	1.78
36	H13	3	5	15	300	0.3	30	1.78
37	H13	3	6	25	100	0.2	45	1.44
38	H13	3	7	25	200	0.1	30	2.25
39	H13	2	6	20	200	0.1	30	1.78
40	H13	2	6	20	200	0.1	15	1.78
41	H13	2	6	20	200	0.1	45	1.78
42	H13	2	6	20	200	0.1	60	1.78
43	H13	2	6	20	200	0.1	75	1.78
44	H13	2	6	20	200	0.1	90	1.78
45	H13	2	2	20	200	0.1	30	1.78
46	H13	2	4	20	200	0.1	30	1.78
47	H13	2	8	20	200	0.1	30	1.78
48	H13	2	10	20	200	0.1	30	1.78
49	H13	2	6	15	200	0.1	30	1.78
50	H13	2	6	25	200	0.1	30	1.78
51	H13	2	6	30	200	0.1	30	1.78
52	H13	2	6	35	200	0.1	30	1.78
53	H13	2	6	20	200	0.2	30	1.78
54	H13	2	6	20	200	0.3	30	1.78
55	H13	2	6	20	200	0.4	30	1.78
56	H13	2	6	20	200	0.5	30	1.78
57	H13	2	6	20	200	0.6	30	1.78
58	H13	2	6	20	200	0.7	30	1.78
59	H13	2	6	20	100	0.1	30	1.78
60	H13	2	6	20	300	0.1	30	1.78
61	H13	1	6	20	300	0.1	30	1.78
62	H13	3	6	20	300	0.1	30	1.78
63	H13	4	6	20	300	0.1	30	1.78
64	H13	2	6	20	200	0.1	30	1.44
65	H13	2	6	20	200	0.1	30	2.25
66	H13	2	6	20	200	0.1	30	2.94
67	AISI52100	2	5	15	100	0.1	15	1.44
68	AISI52100	2	6	25	300	0.1	15	1.78
69	AISI52100	3	5	15	300	0.3	30	1.78
70	AISI52100	3	6	25	100	0.2	45	1.44
71	AISI52100	3	7	25	200	0.1	30	2.25
72	AISI52100	2	6	20	200	0.1	30	1.78
73	AISI52100	2	6	20	200	0.1	15	1.78
74	AISI52100	2	6	20	200	0.1	45	1.78
75	AISI52100	2	6	20	200	0.1	60	1.78
76	AISI52100	2	6	20	200	0.1	75	1.78
77	AISI52100	2	2	20	200	0.1	30	1.78
78	AISI52100	2	4	20	200	0.1	30	1.78
79	AISI52100	2	8	20	200	0.1	30	1.78
80	AISI52100	2	10	20	200	0.1	30	1.78

Table A1. Cont.

Simulation	Material	Ram Speed (mm/s)	Core Diameter (mm)	Billet Height (H)	Temperature (° C)	Friction	Die Semi-Angle (°)	Extrusion Ratio
81	AISI52100	2	6	15	200	0.1	30	1.78
82	AISI52100	2	6	25	200	0.1	30	1.78
83	AISI52100	2	6	30	200	0.1	30	1.78
84	AISI52100	2	6	35	200	0.1	30	1.78
85	AISI52100	2	6	20	200	0.2	30	1.78
86	AISI52100	2	6	20	200	0.3	30	1.78
87	AISI52100	2	6	20	200	0.4	30	1.78
88	AISI52100	2	6	20	200	0.5	30	1.78
89	AISI52100	2	6	20	200	0.6	30	1.78
90	AISI52100	2	6	20	200	0.7	30	1.78
91	AISI52100	2	6	20	100	0.1	30	1.78
92	AISI52100	2	6	20	300	0.1	30	1.78
93	AISI52100	1	6	20	300	0.1	30	1.78
94	AISI52100	3	6	20	300	0.1	30	1.78
95	AISI52100	4	6	20	300	0.1	30	1.78
96	AISI52100	2	6	20	200	0.1	30	1.44
97	AISI52100	2	6	20	200	0.1	30	2.25
98	25CrMo4	2	6	20	200	0.3	45	1.44
99	25CrMo4	2	7	15	200	0.1	45	1.78
100	25CrMo4	3	5	25	200	0.2	15	1.44
101	25CrMo4	3	7	20	100	0.3	15	1.78
102	25CrMo4	2	6	20	300	0.1	30	1.78
103	25CrMo4	2	6	20	200	0.1	30	1.78
104	25CrMo4	2	6	20	200	0.1	15	1.78
105	25CrMo4	2	6	20	200	0.1	45	1.78
106	25CrMo4	2	6	20	200	0.1	60	1.78
107	25CrMo4	2	6	20	200	0.1	75	1.78
108	25CrMo4	2	6	20	200	0.1	90	1.78
109	25CrMo4	2	2	20	200	0.1	30	1.78
110	25CrMo4	2	4	20	200	0.1	30	1.78
111	25CrMo4	2	8	20	200	0.1	30	1.78
112	25CrMo4	2	10	20	200	0.1	30	1.78
113	25CrMo4	2	6	15	200	0.1	30	1.78
114	25CrMo4	2	6	25	200	0.1	30	1.78
115	25CrMo4	2	6	30	200	0.1	30	1.78
116	25CrMo4	2	6	35	200	0.1	30	1.78
117	25CrMo4	2	6	20	200	0.2	30	1.78
118	25CrMo4	2	6	20	200	0.3	30	1.78
119	25CrMo4	2	6	20	200	0.4	30	1.78
120	25CrMo4	2	6	20	200	0.5	30	1.78
121	25CrMo4	2	6	20	200	0.6	30	1.78
122	25CrMo4	2	6	20	200	0.7	30	1.78
123	25CrMo4	2	6	20	100	0.1	30	1.78
124	25CrMo4	2	6	20	300	0.1	30	1.78
125	25CrMo4	1	6	20	300	0.1	30	1.78
126	25CrMo4	3	6	20	300	0.1	30	1.78
127	25CrMo4	4	6	20	300	0.1	30	1.78
128	25CrMo4	2	6	20	200	0.1	30	1.44
129	25CrMo4	2	6	20	200	0.1	30	2.25
130	AISI3310	2	6	20	200	0.1	30	1.78
131	AISI3310	2	6	20	200	0.1	15	1.78
132	AISI3310	2	6	20	200	0.1	45	1.78
133	AISI3310	2	6	20	200	0.1	60	1.78
134	AISI3310	2	6	20	200	0.1	75	1.78
135	AISI3310	2	6	20	200	0.1	90	1.78
136	AISI3310	2	2	20	200	0.1	30	1.78

Table A1. Cont.

Simulation	Material	Ram Speed (mm/s)	Core Diameter (mm)	Billet Height (H)	Temperature (° C)	Friction	Die Semi-Angle (°)	Extrusion Ratio
137	AISI3310	2	4	20	200	0.1	30	1.78
138	AISI3310	2	8	20	200	0.1	30	1.78
139	AISI3310	2	10	20	200	0.1	30	1.78
140	AISI3310	2	6	15	200	0.1	30	1.78
141	AISI3310	2	6	25	200	0.1	30	1.78
142	AISI3310	2	6	30	200	0.1	30	1.78
143	AISI3310	2	6	35	200	0.1	30	1.78
144	AISI3310	2	6	20	200	0.2	30	1.78
145	AISI3310	2	6	20	200	0.3	30	1.78
146	AISI3310	2	6	20	200	0.4	30	1.78
147	AISI3310	2	6	20	200	0.5	30	1.78
148	AISI3310	2	6	20	200	0.6	30	1.78
149	AISI3310	2	6	20	200	0.7	30	1.78
150	AISI3310	2	6	20	100	0.1	30	1.78
151	AISI3310	2	6	20	300	0.1	30	1.78
152	AISI3310	1	6	20	300	0.1	30	1.78
153	AISI3310	3	6	20	300	0.1	30	1.78
154	AISI3310	4	6	20	300	0.1	30	1.78
155	AISI3310	2	6	20	200	0.1	30	1.44
156	AISI3310	2	6	20	200	0.1	30	2.25
157	AISI3310	2	6	20	200	0.1	30	2.94

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