

Creative

Holistic Approach on the Research of Yielding, Creep and Fatigue Crack Growth Rate of Metals Based on Simplified Model of Dislocation Group Dynamics

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Abstract: The simplified model of numerical analyses of discrete dislocation motion and emission from a stressed source was applied to predict the yield stress, dislocation creep, and fatigue crack growth rate of metals dominated by dislocation motion. The results obtained by these numerical analyses enabled us to link various dynamical effects on the yield stress, dislocation creep, and fatigue crack growth rate with the experimental results of macroscopic phenomena, as well as to link them with theoretical results obtained by the concept of static, continuously distributed infinitesimal dislocations for the equilibrium state under low strain or stress rate conditions. This will be useful to holistic research approaches with concern for time and space scales, that is, in a time scale ranging from results under high strain rate condition to those under static or low strain rate condition, and in a space scale ranging from meso-scale to macro-scale mechanics. The originality of results obtained by these analyses were found by deriving the analytical formulations of number of dislocation emitted from a stressed source and a local dynamic stress intensity factor at the pile-up site of dislocations as a function of applied stress or stress rate and temperature material constants. This enabled us to develop the predictive law of yield stress, creep deformation rate, and fatigue crack growth rate of metals dominated by dislocation motion. Especially, yielding phenomena such as the stress rate and grain size dependence of yield stress and the delayed time of yielding were clarified as a holistic phenomenon composed of sequential processes of dislocation release from a solute atom, dislocation group moving, and stress concentration by pile-up at the grain boundary.

Keywords: holistic approach; dislocation group dynamics; dynamic factor; dislocation pile-up; yield stress; dislocation creep; fatigue crack growth rate

1. Introduction

The purpose of the research of dislocation mechanics is considered to have two directionalities, that is, application to the research of materials science and application to the research of the strength of materials such as that of yielding and fatigue crack growth rate.

The former closely relates to micro plasticity, such as the conditions of dislocation emission, annihilation, and cross slip, and this research was developed in the manner of the modern dislocation dynamics [1].

The latter closely relates to connect problems of the strength and fracture of materials with the macro scale [2–5].

This article is related to the fracture of materials. For this case, the numerical results of the number of moving dislocations emitted from a stressed source and local stress concentration caused by a dislocation pile-up were necessary to be formulated as an analytical function of applied stress or stress rate, temperature, and material constants [2,3].

Since the behaviors of dislocation group motion have a scale of μm at the meso-scale level, that is, the intermediate scale of nm (an individual dislocation) and mm (crack length) scales, which is a comparable scale with grain size. At this scale, the simplification of the model of analysis is considered to be more convenient to derive a predictive law of the strength and fracture of metals.

To fulfil this purpose, in our research, the establishment of a predictive theory of strength and fracture of materials was conducted by conducting the simplification of the model of analysis [3] and verification with experimental results [3,6].

Yielding phenomena such as stress rate and grain size dependence of yield stress and delayed time of yielding were especially clarified as a holistic phenomenon composed of sequential processes of dislocation release from a solute atom [7,8], dislocation group moving [6], and stress concentration by a pile-up at the grain boundary [2–4].

2. Dislocation Groups Dynamics Aimed for Applications to Problems of Yielding, Creep, and Fatigue [2,9,10]

2.1. Model, Basic Equation and Method of Analysis [2,9,10]

Plastic deformation is caused by dislocation group motion emitted from a stressed source.

It closely relates to plastic yielding and fatigue crack growth dominated by discrete dislocations emitted from a stressed source near a crack tip. Pioneering works on the analysis of discrete dislocation group dynamics emitted from a stressed source have been conducted [4,11,12], but a power law equation has been adopted between the isolated dislocation velocity and the stress for practical application to the strength of materials such as the yield stress, creep rate, and fatigue crack growth rate; this is given by Equation (1), which is related to experimental equations [4].

Equations of discrete moving dislocation groups are given by Equations (1) and (2).

$$V_i = \frac{dx_i}{dt} = M\tau_{eff,i}^m \quad (1)$$

$$\tau_{eff,i} = \dot{\tau}t + A \sum_{\substack{j=1 \\ i \neq j}}^n \frac{1}{x_i - x_j} \quad (i = 1 \sim n) \quad (2)$$

In the equations, V_i and x_i are velocity and position of individual dislocations in a linear array, respectively, and M and m are the material constants of an isolated dislocation given by the experimental Equation (3) [13]. The calculation starts at the time of $t = 0.0$ s.

$$v = v_0 \left(\frac{\tau}{\tau_0^*} \right)^m V \quad (3)$$

In Equation (3), $\dot{\tau}$ is the stress rate; t is the time of stress application; $\tau_{eff,i}$ is the effective stress exerted on individual dislocations in terms of shear stress; and $A = \frac{Gb}{2\pi(1-\nu)}$, where G is the shear modulus, b is the Burgers vector, ν is the Poisson's ratio, $V_0 = 1$ cm/s, and τ_0^* is a constant representing the stress required to give a dislocation velocity $v = 1$ cm/s (resistant stress against the dislocation motion).

A free expansion model of linear dislocation motion emitted from a stressed source, S , is shown in Figure 1 [9,10].

The numerical analyses were conducted as follows [9].

When the effective stress exerted on dislocation source ($x = 0.0$) takes the source activation stress, the new dislocation is originated at $x = 0.0$, and these processes are iterated. Equations (2) and (1) were

solved by the Runge–Kutta Merson method. The effective stress exerted on a dislocation source is given by Equation (4).

$$\tau = \dot{\tau} t$$

$$\tau_{eff,s} = \dot{\tau} t - A \sum_{j=1}^n \frac{1}{x_j} \quad (4)$$

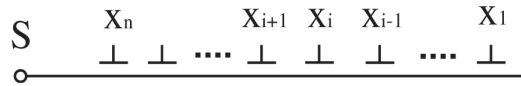


Figure 1. Free expansion model of linear dislocation motion emitted from a stressed source, S.

2.2. Discrete Dislocation Groups Dynamics of Free Expansion and Similarity Law of Dislocation Flow [9,10]

From the analysis, the ratio of positions, velocity and effective stress of individual dislocation in the dislocation array to those of an isolated dislocation, such as $\frac{x_i}{x_{iso}}$, $\frac{v_i}{v_{iso}}$, and $\frac{\tau_{effi}}{\tau_{iso}}$, were found to be dominated by $\Theta = \left(\frac{\dot{\tau}}{\dot{\tau}_0}\right)^{\frac{(m+1)}{(m+2)}} \theta$, which is named the dynamic factor [9,10]. Here, τ is the applied stress acting on dislocations in the dislocation array, θ is the non-dimensional time controlled by t_0 , and t_0 is the time of an isolated dislocation moving the distance, l . It is given by $t_0 = \left[\frac{(m+1)l}{M\dot{\tau}_0^m}\right]^{\frac{1}{(m+1)}}$ [2,9,10] by using Equation (3). In this analysis, l was taken as the length of 0.01 mm.

Furthermore, the number of dislocations emitted from a stressed source were also found to be given by the dynamic factor, Θ , which is a non-dimensional character, as shown in Equation (5a); the dimensional parameter η is shown in Equation (5b) [9,10].

$$N = A\Theta^{m+1} \quad (5a)$$

$$N = A_0\eta^{m+1} \quad (5b)$$

$$\eta = \dot{\tau}^{\frac{m+1}{m+2}} t = \tau \dot{\tau}^{-\frac{1}{m+2}}$$

In the equation, A and A_0 are non-dimensional and dimensional constants, respectively. By dimensional analysis, N is given by Equation (6a) [9,10].

$$N = \gamma(m) \left(\frac{M}{Gb}\right)^{\frac{(m+1)}{(m+2)}} \eta^{m+1} \quad (6a)$$

$$\gamma(m) = 1.4m^{-1.45} \quad (6b)$$

In these equations, $\gamma(m)$ is a non-dimensional function depending on m .

The velocity of an isolated dislocation is given by thermally activated process, as shown in Equation (2) [14].

$$v = A_1 \exp\left(-\frac{H}{RT}\right) \quad (7)$$

$$H = H_k \left(1 + \frac{1}{4} \ln \frac{16\tau_p^0}{\pi\tau}\right) \quad (8)$$

In these equations, A_1 is a constant, H_k is the kink energy, τ is the applied stress, τ_p^0 is the Peierls stress at 0° K, and T is the absolute temperature. By substituting Equation (8) into Equation (7) and comparing Equation (7) with Equation (9), which is the experimental equation of an isolated dislocation, Equations (10a) and (10d) could be obtained.

$$v = M\tau^m \quad (9)$$

$$m = \frac{H_k}{4kT} \quad (10a)$$

$$M = v_0 \left(\frac{1}{\tau_0^*} \right)^m \quad (10b)$$

$$\tau_0^* = \tau_{00} \left(\frac{A_1}{v_0} \right)^{-1/m} \quad (10c)$$

$$\tau_{00} = e^4 \left(\frac{16}{\pi} \right) \tau_P^0 \quad (10d)$$

Using Equations (10a)–(10d), Equation (6a) was able to be rewritten as Equation (11). From this equation, the number of dislocations emitted from a stressed source was found to be dominated by a thermally activated process [10].

$$N = A_* t_a^{\frac{m+1}{m+2}} \left(\frac{\tau}{G} \right)^{\frac{m+1}{m+2}} \exp \left\{ -\frac{m+1}{m+2} H_k \ln \left(\frac{\tau_{00}}{\tau} \right) / 4kT \right\} \quad (11)$$

$$\text{where} \quad A_* = \gamma(m) \left(\frac{b}{A_1} \right)^{-\frac{m+1}{m+2}} \quad (12)$$

2.3. Dislocation Pile-Up Induced by Local Stress Field [2]

Some previous research has treated analyses of dislocation pile-up [4,11,15], but there has not been so much research that has considered the application to fracture mechanics description.

In this section, numerical analyses were conducted on the dynamic piling-up of discrete dislocations emitted from a stressed source and on the dynamic stress intensity factor caused by discrete moving dislocations in a pile-up.

2.3.1. Model, Basic Equation and Analysis [2]

Until the lead dislocation in the array arrives at a barrier, such as grain boundary, dislocations will emit from a stressed source and move freely except for the interactions between dislocations, as shown in Figure 1. Equations of the motion of dislocation groups are given by Equations (1) and (2). After the arrival of the lead dislocation at the barrier, it is locked and the trailing dislocations pile-up against the barrier, as shown in Figure 2.

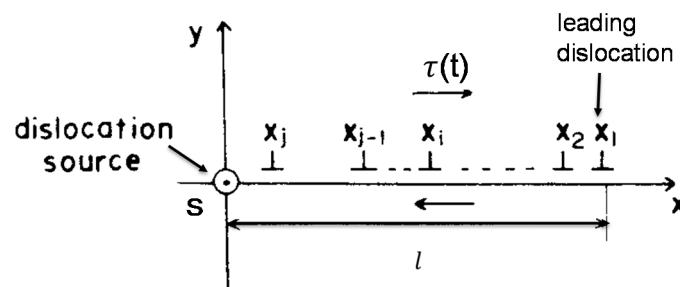


Figure 2. Discrete dislocation pile-up model with emission from a stressed source.

Equations of effective stress exerted on each dislocation in the array for the case of dislocation group pile-up are given by Equation (13).

$$\tau_{eff,i} = \tau t + A \left(\frac{1}{x_i - l} + \sum_{\substack{j=2 \\ i \neq j}}^n \frac{1}{x_i - x_j} \right) \quad i = 1 \sim n \quad (13)$$

The equation of motion of each dislocation in the array is calculated by Equation (1).

The effective stress exerted on a dislocation source before and after the leading dislocation arrives at the site of pile-up is given by Equations (14a) and (14b), respectively.

$$\tau_{eff,s} = \dot{\tau}t - A \sum_{j=2}^n \frac{1}{x_j}, \quad (x_1 < l) \quad (14a)$$

$$\tau_{eff,s} = \dot{\tau}t - A \left(\frac{1}{l} + \sum_{j=2}^n \frac{1}{x_j} \right), \quad (x_1 = l) \quad (14b)$$

The stress distribution, $\tau(x, t)$, caused by dynamical piling up in the region of $x > l$ is shown in Figure 3 and is given by Equation (15).

$$\tau(x, t) = \frac{A}{x-l} + A \sum_{i=2}^n \frac{1}{x-x_i} + \dot{\tau}t \quad (15)$$

The dynamic stress intensity factor caused by dislocation pile-up formation is given by Equations (2)–(16).

$$k(t) \cong \sqrt{2\pi(x-l)\tau(x, t)_{l < x < l(1+\epsilon)}} \quad (16)$$

where $l\epsilon$ is the small distance in which the stress distribution has the characteristic of

$$\frac{1}{\sqrt{x^*}} \quad (x^* = x - l = l\epsilon)$$

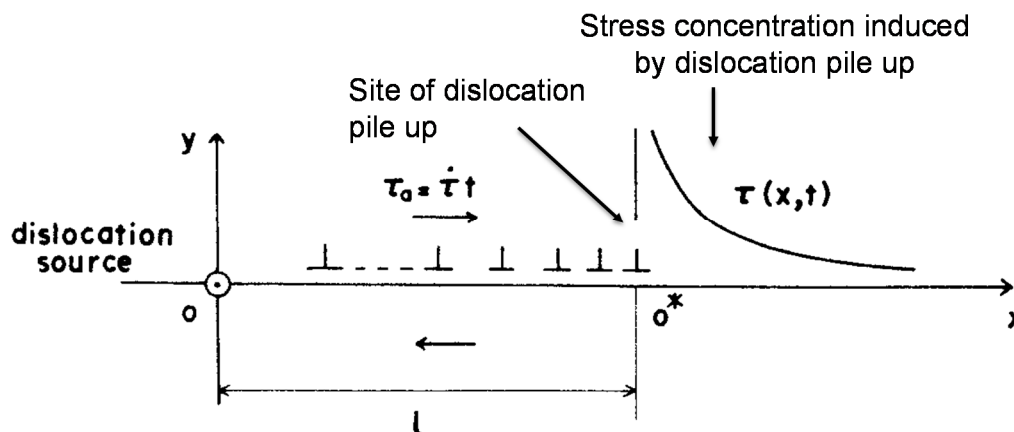


Figure 3. The stress distribution, $\tau(x, t)$, caused by dynamical piling up [3].

2.3.2. Results [2,3]

The numerical results of stress distribution on the slip line $x > l$ near the site of pile-up (barrier) were obtained by Equation (15), as shown in Figures 4 and 5, where $s^* = s - 1$ is the non-dimensional distance from the site of dislocation pile-up (O^* in Figure 3), S is the non-dimensional value of x controlled by l .

When the number of dislocations emitted is as small as shown in Figure 4, the stress distribution near the barrier shows a $1/s^*$ singularity, and with an increase in s^* , a $\frac{1}{\sqrt{s^*}}$ singularity appears but is restricted within a narrow region. ($3 \times 10^{-3} < s^* < 2 \times 10^{-2}$) The characteristics of $\frac{1}{\sqrt{s^*}}$ take minor portion in the stress distribution [2].

On the other hand, when the number of emitted dislocations increases, as shown in Figure 5 (sixty emitted dislocations), the stress distribution shows a singularity of $\frac{1}{\sqrt{S^*}}$, which appears from the vicinity of the barrier; this characteristic region extends up to 5% of the length of slip line [2].

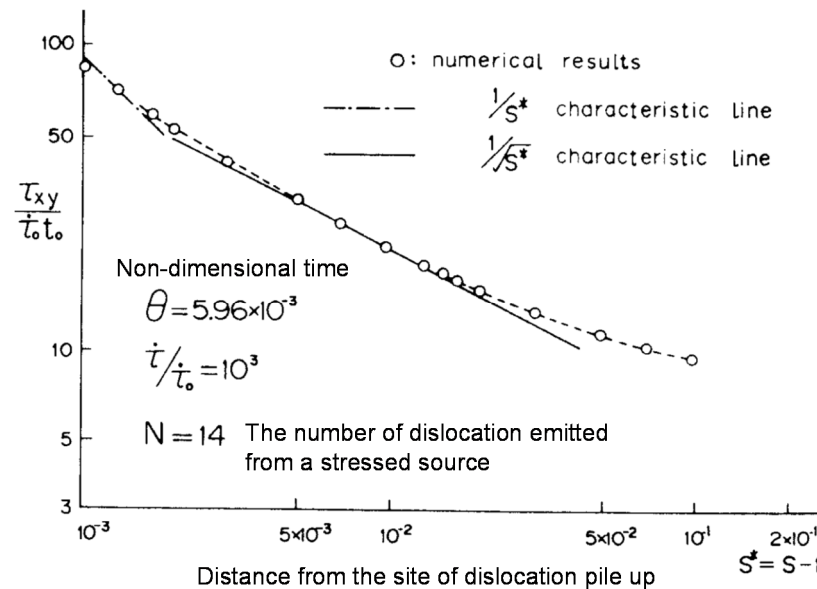


Figure 4. The numerical results of stress distribution on the slip line, $x > l(S^*)$, near the site of pile-up (barrier) obtained by Equation (15) for the case of $N = 14$ (Small number of dislocations) [2].

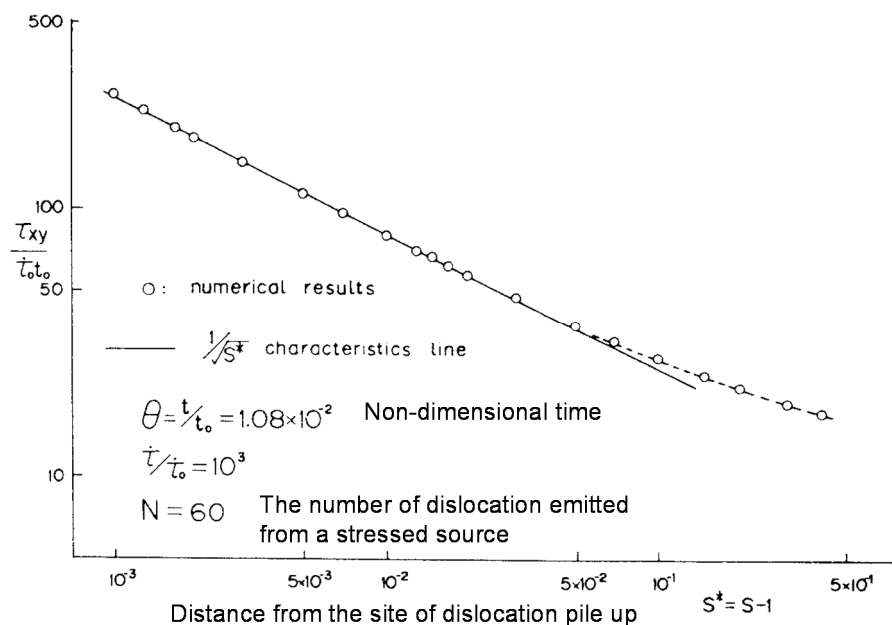


Figure 5. The numerical results of stress distribution on the slip line, $x > l(S^*)$, near the site of pile-up (barrier) obtained by Equation (15) for the case of $N = 60$ (large number of dislocations) [2].

The numerical results of the dynamic stress intensity factor $\overline{K_d}(\theta)$ due to pile-up by discrete dislocation groups dynamics with emission were obtained, as shown in Figure 6, by using Equations (15) and (16) and by the $\frac{1}{\sqrt{S^*}}$ singularity of the stress distribution, as in Figure 5. A static solution, $\overline{K_s}(\theta)$, obtained by the concept of continuously distributed infinitesimal dislocations for the equilibrium

pile-up is given by Equation (17) and is also shown in Figure 6 for comparison with the dynamic stress intensity factor, $\overline{K}_d(\theta)$. $\overline{K}_s(\theta)$ and the linear part of $\overline{K}_d(\theta)$ in Figure 6 are written as follows.

$$\overline{K}_s(\theta) = 10^3 \theta \quad (17)$$

$$\overline{K}_d(\theta) = 1.2 \times 10^3 \theta - 5.2 \quad (18)$$

$$\text{where } \theta = \frac{t}{t_0}$$

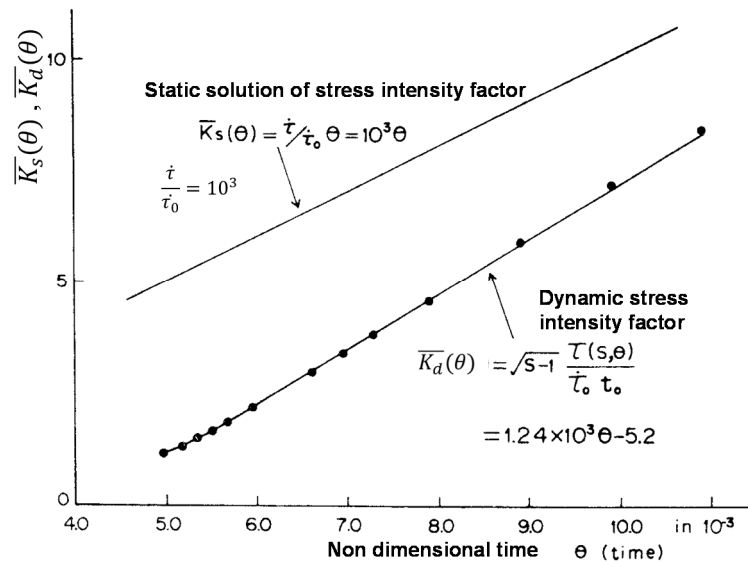


Figure 6. Numerical results of the dynamic stress intensity factor, $\overline{K}_d(\theta)$, due to pile-up by dislocation group dynamics with emission, as well as a comparison with that of a static solution [2].

The values of $\overline{K}_d(\theta)/\overline{K}_s(\theta)$ were plotted against non-dimensional time, θ , as shown in Figure 7. From these results, it can be seen that the dynamic stress intensity factor is smaller than the static one and asymptotically approaches the static one as the number of emitted dislocations increases.

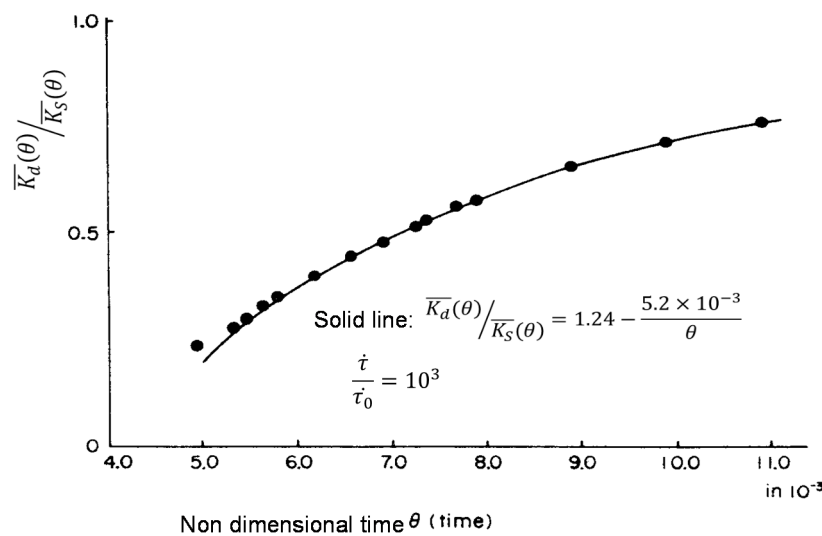


Figure 7. The values of $\overline{K}_d(\theta)/\overline{K}_s(\theta)$ plotted against non-dimensional time, θ [2].

The calculated dynamic dislocation density distribution f_d is shown in Figures 8–11 [3]. On the other hand, the static dislocation density distribution f_s and the number of dislocations N under

equilibrium state without emitting are given by the continuous distributed infinitesimal dislocations concept, assuming $f_s(s) = 0$ at the dislocation source and $f_s(s) \rightarrow \infty$ at the site of pile-up, respectively. They are given by Equations (19) and (20).

$$f_s(s) = \frac{\tau_a}{\pi A^*} \sqrt{\frac{s}{1-s}} \quad (19)$$

$$N = \int_0^d f_s(s) ds = \frac{\tau_a d}{2A^*} \quad (20)$$

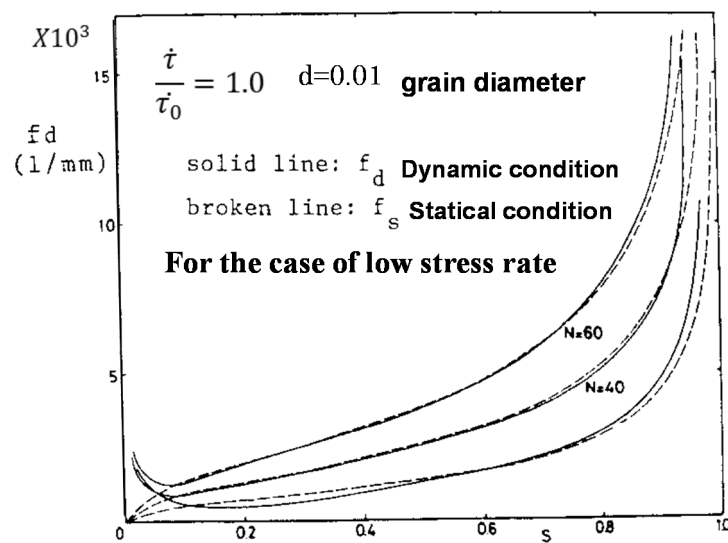


Figure 8. The dynamic dislocation density distribution f_d for grain size (d) = 0.01; non-dimensional stress rate $\left(\frac{\dot{\tau}}{\tau_0}\right) = 10$ [3].

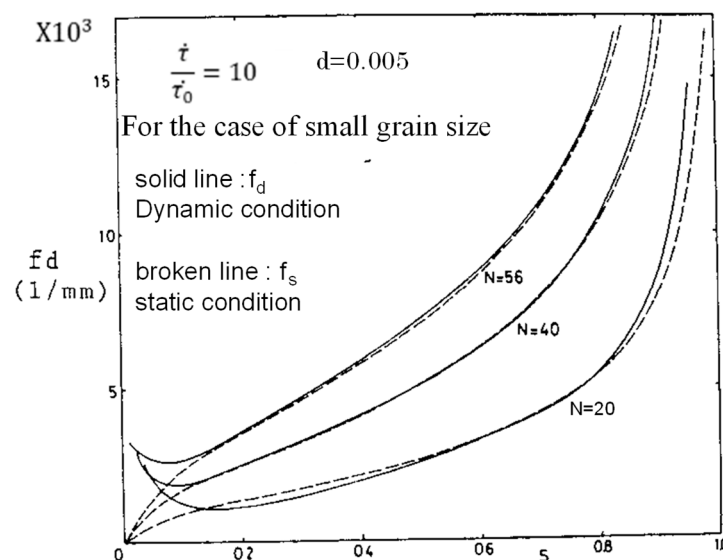


Figure 9. The dynamic dislocation density distribution f_d for $d = 0.005$; $\frac{\dot{\tau}}{\tau_0} = 10$ [3].

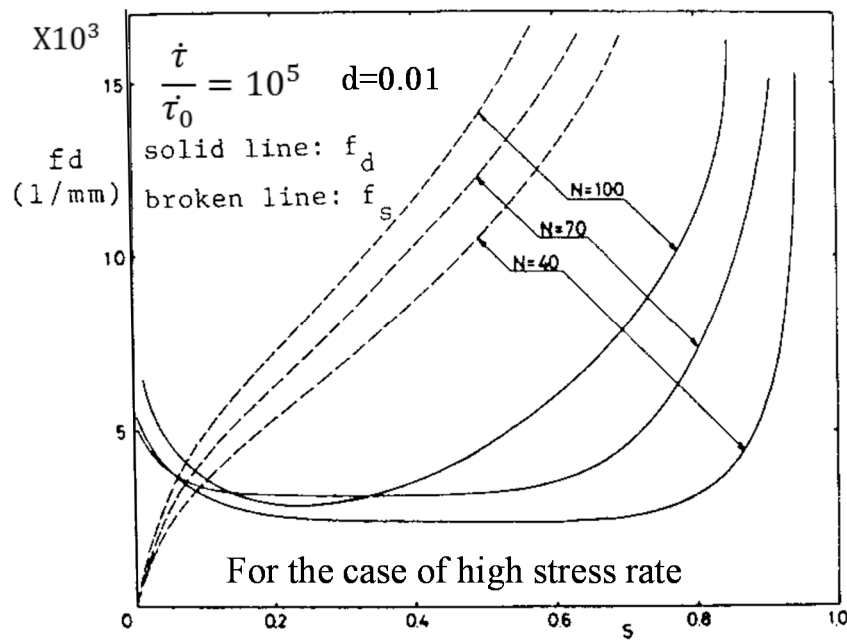


Figure 10. The dynamic dislocation density distribution f_d for $d = 0.01$; $\frac{\dot{\tau}}{\tau_0} = 10^5$ [3].

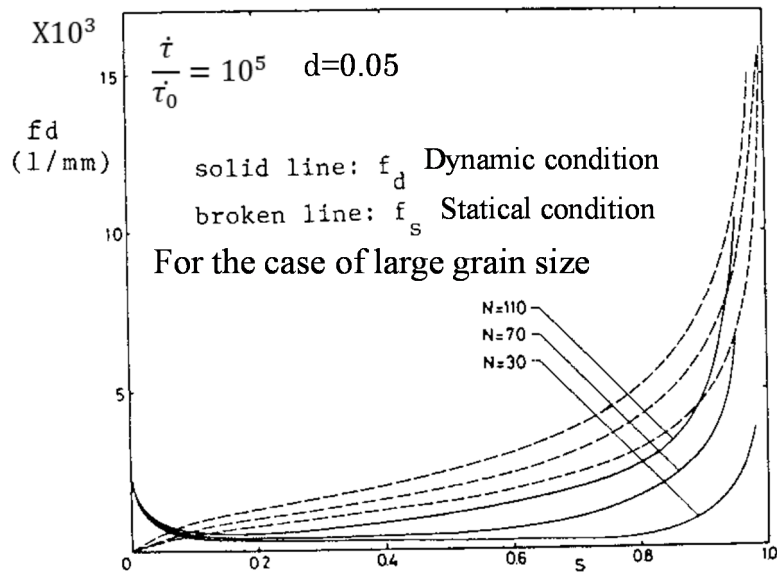


Figure 11. The dynamic dislocation density distribution f_d for $d = 0.05$; $\frac{\dot{\tau}}{\tau_0} = 10$ [3].

In Figures 8–11, f_s is shown by the dotted line. These results showed when the stress application rate, $\dot{\tau}$, or grain size, d , is small, and a static equilibrium solution based on a continuous distribution of infinitesimal dislocations gives a good approximation, as shown in Figures 8 and 9. However, with increase in $\dot{\tau}$ or d , the dynamic effect becomes more remarkable and f_d becomes smaller than f_s [3], as shown in Figures 10 and 11 [3].

The dynamic stress intensity factor $\overline{K_d}(\theta)$ at the site of pile-up such as grain boundary in non-dimensional form can be obtained using Equation (16). In Figure 12, $\overline{K_d}(\theta)/\overline{K_s}(\theta)$ is plotted against the non-dimensional dynamic factor $\Theta = \left(\frac{\dot{\tau}}{\tau_0}\right)^{\frac{(m+1)}{(m+2)}} \theta$ [9,10]. For the case of iron ($m = 3$ [13]), the following equation was obtained [3].

$$\frac{K_d}{K_s} = 1.0 - A \exp(-B\Theta) \quad (21)$$

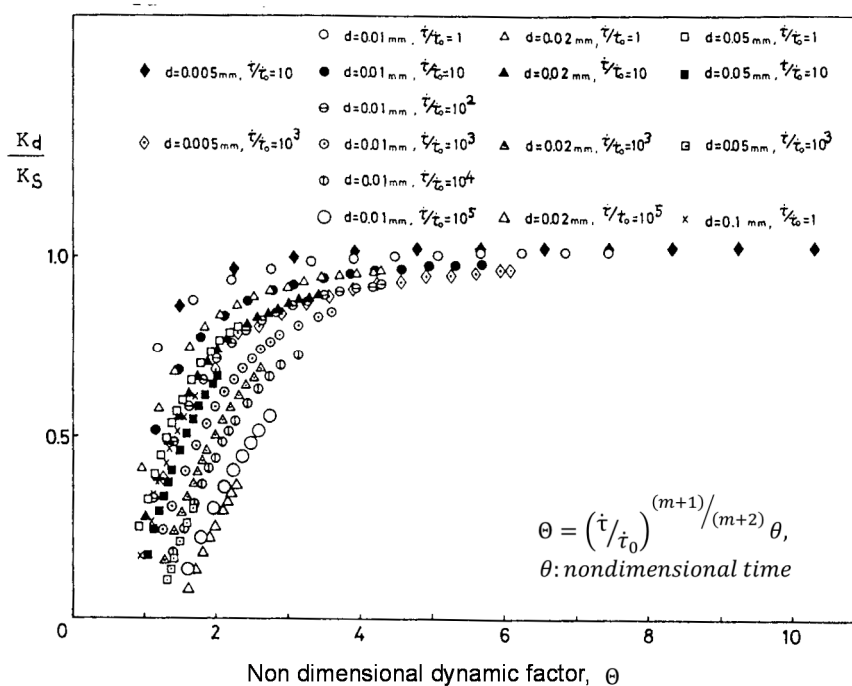


Figure 12. The relationship between dynamic stress intensity factor controlled by static stress intensity factor and the non-dimensional dynamic factor Θ [3].

By using the non-linear least square method, A and B in Equation (21) were obtained as a function of $\frac{\dot{\tau}}{\tau_0}$ and d, respectively, as follows [3].

$$A = \varphi\left(\frac{\dot{\tau}}{\tau_0}, d\right) = 2.35 - 0.0214\left(\frac{\dot{\tau}}{\tau_0}\right)^{-0.172} d^{-0.896} \quad (22)$$

$$B = \psi\left(\frac{\dot{\tau}}{\tau_0}, d\right) = 1.08\left(\frac{\dot{\tau}}{\tau_0}\right)^{-0.0557} \quad (23)$$

The results of Figures 13 and 14 show that Equations (22) and (23) well-represent numerical results.

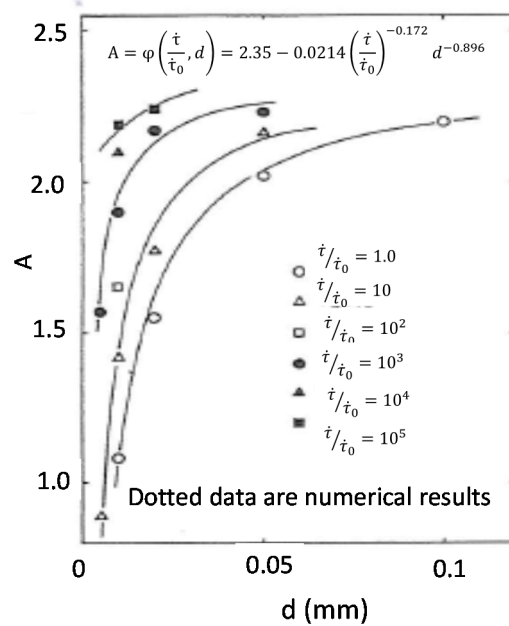


Figure 13. The relationship between a non-dimensional constant (A) and d [3].

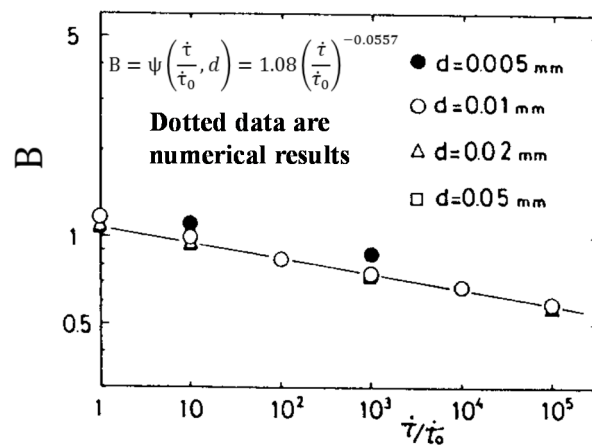


Figure 14. The relationship between B and $\frac{\dot{\tau}}{\dot{\tau}_0}$ [3].

By substituting Equations (22) and (23) into Equation (21), Equation (24) can be obtained [3].

$$\frac{K_d}{K_s} = 1.0 - \left\{ 2.35 - 0.0214 \left(\frac{\dot{\tau}}{\dot{\tau}_0} \right)^{-0.172} d^{-0.896} \right\} \exp \left\{ -1.08 \left(\frac{\dot{\tau}}{\dot{\tau}_0} \right)^{0.75} \frac{t}{t_0} \right\} \quad (24)$$

Figure 15 shows that the calculated values from Equation (24) were found to be in good agreement with data obtained by numerical analyses.

By using dimensional analysis and determining the coefficient of the constant term using the number of material constants, Equation (24) leads to Equation (25) [3].

$$K_d(t) = \sqrt{2\pi d} (\dot{\tau} t) \left[1.0 - \left\{ 2.35 - 0.230 \left(\frac{G}{\tau_0^*} \right)^{\frac{m}{(m+2)}} \left(\frac{GV_0}{\dot{\tau} b} \right)^{\frac{1}{(m+2)}} \left(\frac{b}{d} \right)^{\frac{(m+1)}{(m+2)}} \right\} \right. \\ \left. \times \exp \left\{ -1.08 \left(\frac{\dot{\tau} t}{\tau_0^*} \right)^{\frac{m}{(m+2)}} \left(\frac{V_0 t}{(m+1)d} \right)^{\frac{1}{(m+1)}} \right\} \right] \quad (25)$$

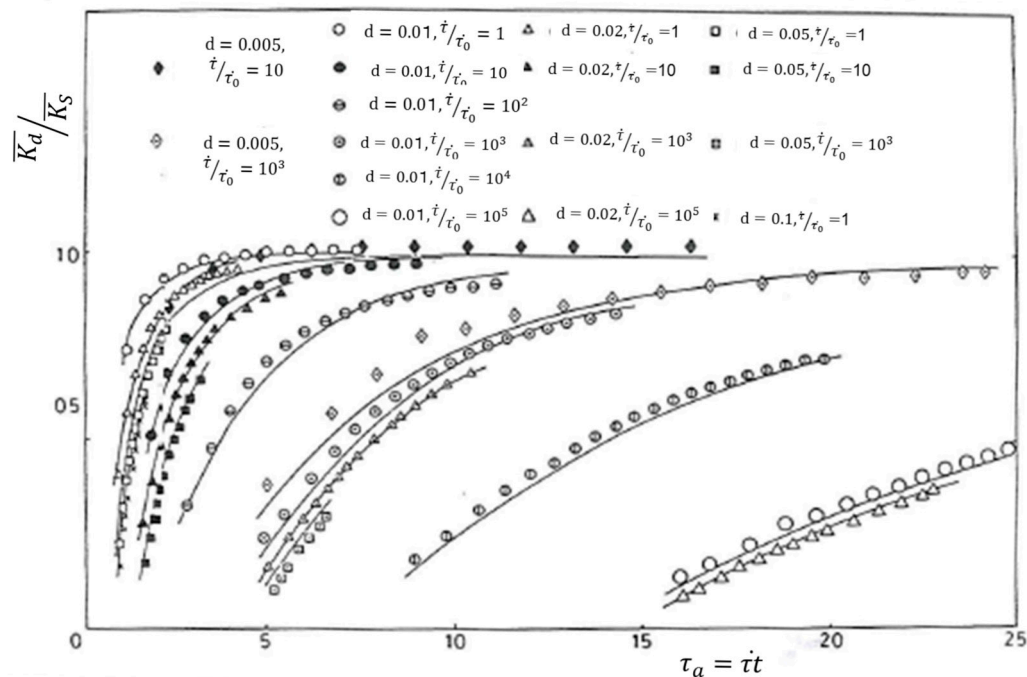


Figure 15. The relationship between dynamic stress intensity factor controlled by static stress intensity and applied stress under constant stress rate condition [3].

2.4. Application to Problem of Yielding

2.4.1. Basic Equations

Cottrell and Bilby described the mechanism of yielding from the view point of dislocation trapping mechanism by solute atoms of carbon [7]; however, the effect of strain rate and temperature have not yet been fully formulized. Concerning ductile fracture of steels, the formulation of the upper yielding point and delay time for yielding are important factors to be analyzed. Takeo Yokobori conducted the formulation of the stress rate dependence of the upper yield point based on a stochastic model analysis that analyzed the releasing process of solute atom from a dislocation [8].

On the other hand, on the basis of the concept that the velocity of an isolated dislocation is proportional to the strain rate of a specimen given by Equation (26), Johnson [15] and Hahn [16] described the yielding phenomenon from the view point of theory of dislocation [15,16]. However, a cleared formulation of yielding phenomenon including the effect of temperature has not yet been conducted. Furthermore, the researchers used the equation of velocity of an isolated dislocation motion by considering that every dislocation moves at the same velocity without interaction between them, as given by Equation (26) [16].

$$\dot{\gamma} = \rho b \bar{v} \quad (26)$$

In the equation, ρ is the dislocation density, b is the Burgers vector, and \bar{v} is the mean velocity of a dislocation in which the equation of velocity of an isolated dislocation was used.

In this section, instead of Equation (26), Equation (27) [4,17], which considers the interaction of dislocations within groups starting with dislocation emission from a stressed source (as calculated by Equations (1) and (2)) was adopted.

$$\dot{\gamma} = \rho b \sum_{i=1}^{n(t)} v_i \quad (27)$$

In Equation (27), $n(t)$ is the number of dislocations emitted from a stressed source at the time of t , i is the dislocation number, and v_i is the velocity of the i th dislocation in the dislocation groups.

By conducting computer simulation using the physical model of Figure 1 and Equations (1)–(4), the summation of non-dimensional velocity of each dislocation in the array was found to be written by Equation (28) and is shown in Figure 16 [17].

$$\sum_{i=1}^n \frac{v_i(t)}{v_{iso}} \cong n(t) \quad (28)$$

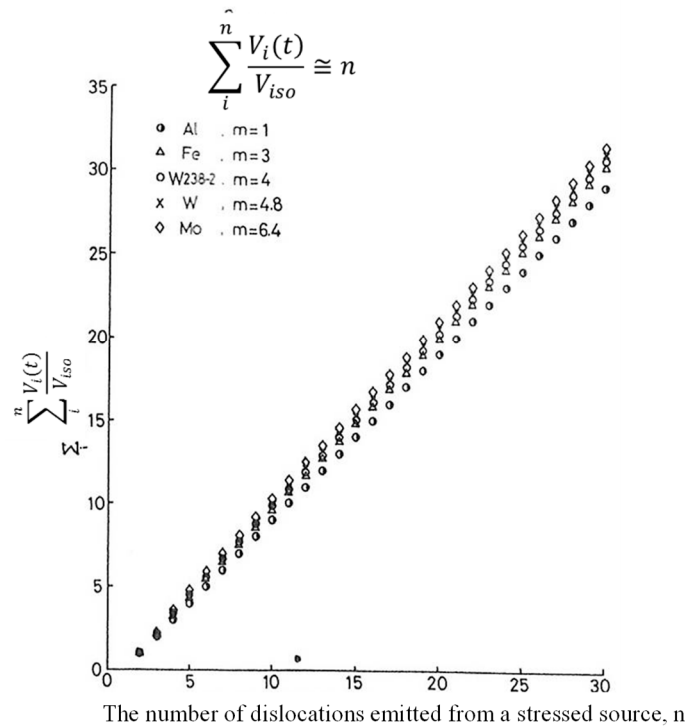


Figure 16. The relationship between $\sum_{i=1}^n \frac{v_i(t)}{v_{iso}}$ and the number of dislocation emitted from a stressed source, n [17].

By substituting Equation (28) into Equation (27) and using Equation (29), the plastic strain rate is given by Equation (30) [17].

$$\rho = \rho_0^* \Lambda^* \quad (29)$$

In Equation (29), ρ is the area density dislocation, ρ_0^* is the volume density dislocation, and Λ^* is the average length of dislocation.

$$\dot{\gamma} = \rho_0^* \Lambda^* b n(t) v_0 \left(\frac{\tau}{\tau_0} \right)^m = \bar{\rho}(t) b v \quad (30)$$

In Equation (30), $\bar{\rho}(t) = \rho_0^* \Lambda^* n(t)$, and v is given by Equation (3).

2.4.2. The Application of This Theory to Yielding of Steels

The Delay Time of Yielding

The delay time of yielding under rapid application of constant stress is calculated by Equation (31) as the time of plastic strain, thus taking the specified value [16].

$$\gamma_P = \int_0^t \dot{\gamma} dt = \text{const} \quad (31)$$

Using Equations (30) and (32), which are the number of dislocations emitted from a stressed source under constant stress condition [6], and by substituting Equations (30) and (32) into Equation (31) and integrating Equation (31), Equation (33) was obtained [17].

$$n(t) = 2.45m^{-0.865} \left(\frac{b}{v_0 t} \right)^{-\frac{(m+1)}{(m+2)}} \left(\frac{\tau_0^*}{G} \right)^{-\frac{m(m+1)}{(m+2)}} \left(\frac{\tau_a}{G} \right)^{\frac{(m+1)^2}{(m+2)}} \quad (32)$$

where G is the shear modulus. $\tau_a = \tau_Y = \frac{\sigma_Y}{2}$, where σ_Y is the yield stress under uniaxial tensile load, is assumed.

$$t = \left(\frac{\gamma_P}{\rho_0^* \Lambda b^2} \frac{2m+3}{m+2} \frac{m^{0.865}}{2.45} \right)^{\frac{m+2}{2m+3}} \frac{b}{v_0} \left(\frac{G}{\tau_0^*} \right)^{\frac{m+1}{2m+3}} \left(\frac{\sigma_Y}{2\tau_0^*} \right)^{-\frac{(2m^2+4m+1)}{(2m+3)}} \quad (33)$$

By using Equations (7)–(10), Equation (33) was able to be written in the following manner as a function of yield stress, temperature, and material constants [17].

$$t = \left(\frac{0.792\gamma_P}{\rho_0^* \Lambda b^2} \right)^{0.515} \left(\frac{H_k}{4kT} \right)^{0.446} \frac{b}{A_1} \left(\frac{G}{\tau_{00}} \right)^{0.485} \left(\frac{\sigma_Y}{2\tau_{00}} \right)^{-\left(\frac{H_k}{4kT} \right)} \quad (34)$$

In Equation (34), σ_Y is the yield stress under uniaxial tensile loading. As the criterion of yielding, $\gamma_P = 0.01$ and $\rho_0^* \Lambda^* = 10^8/m^2$ were selected [15], and $G = 79.38$ GPa, $\tau_0^* = 169.6$ MPa [10], $m = 10$ [16] were taken as those under room temperature and $b = 3 \times 10^{-10}m$ was selected. Furthermore, for $m = 10\sim 30$ (for steel) [13], $(2m^2 + 4m + 1)/(2m + 3) \approx m$ was approximately assumed.

A comparison of Equation (34) with experimental data is shown in Figure 17 [17]. Equation (34) was found to well-predict experimental data [18,19]. Furthermore, this equation was in good agreement with that obtained based on dislocation dynamics theory that defined yielding to occur when the dislocation density takes some critical value [6] as follows.

$$t = \left(\frac{\bar{N}_0}{\rho_0^* \Lambda 2.45} \right)^{1.06} \left(\frac{H_k}{4kT} \right)^{0.917} \frac{b}{A_1} \left(\frac{G}{\tau_{00}} \right) \left(\frac{\sigma_Y}{2\tau_{00}} \right)^{-\left(\frac{H_k}{4kT} \right)} \quad (35)$$

In Equation (35), \bar{N}_0 is critical dislocation density at the yielding.

This means that the γ_P criterion is identical to the \bar{N}_0 criterion.

Furthermore, Equation (34) was found to be in good qualitative agreement with theoretical results [8] based on Cottrell-Bilby's dislocation release mechanism [7] for locking by solute atoms such as carbon or nitrogen, as given by Equation (36). This means that the locking mechanism closely connects with the mechanism of dislocation group dynamics, as described in the following expression.

$$t = t_0 \left(\frac{\sigma_Y}{\sigma_0} \right)^{-\frac{1}{nkT}} \quad (36)$$

where t_0 and σ_0 are material constants.

T. Yokobori found that adopting a friction stress, τ_i , to resist the motion of a dislocation in Equation (36) was very effective in obtaining agreement with experimental results via equations [20]. In this theory, the effect of τ_i on delay time for yielding, as included in Equation (33), was extended to then give Equation (37) [17].

$$t = \left(\frac{\gamma_P}{\rho_0^* \Lambda b^2} \frac{2m+3}{m+2} \frac{m^{0.865}}{2.45} \right)^{\frac{m+2}{2m+3}} \frac{b}{v_0} \left(\frac{G}{\tau_0^* - \tau_i} \right)^{\frac{m+1}{2m+3}} \left(\frac{\sigma_Y}{2(\tau_0^* - \tau_i)} \right)^{-\frac{(2m^2+4m+1)}{(2m+3)}} \quad (37)$$

By adopting $\tau_i = 86.3$ MPa and using Equations (10a)–(10d), Equation (37) was found to well-predict experimental results [19], as shown in Figure 18 [17].

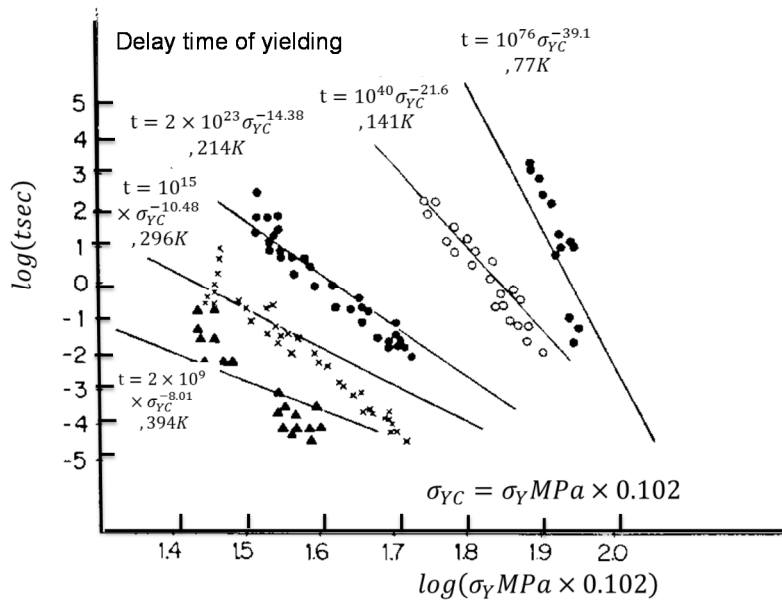


Figure 17. The relationship between the delay time of yielding and yield stress based on dislocation dynamics. Solid lines represent the theoretical results [17]. Dotted lines represent the experimental results [19].

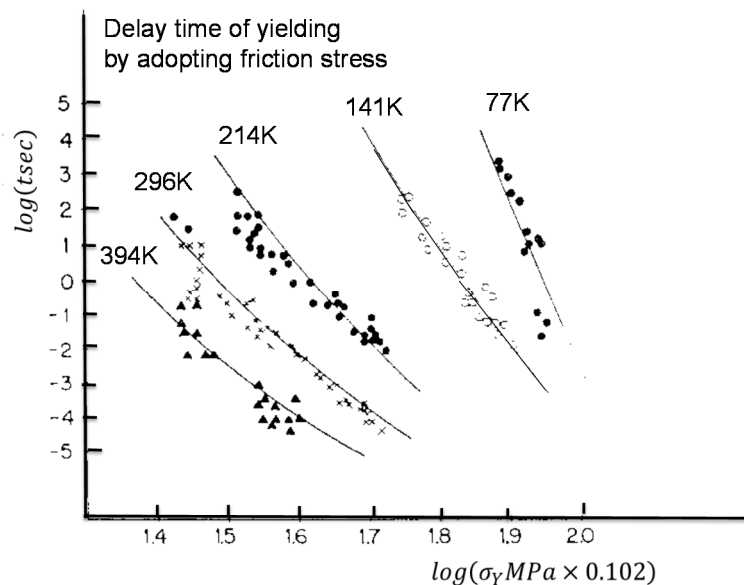


Figure 18. The relationship between the delay time of yielding and yield stress based on dislocation dynamics by accounting for the effect of friction stress of dislocation on delay time. Solid lines represent the theoretical results [17]. Dotted lines represent the experimental results [19].

Furthermore, a previous numerical analysis based on the pile-up behaviors of moving dislocations emitted from a stressed source was conducted [4], and the theoretical relationship between delay time of yielding and yield stress was derived for various grain size [4]. These results were found to be in good agreement with experimental results and were also found to produce similar characteristics to the results given by Figure 18 [17].

From the results mentioned above, the criteria for the various cases of dislocation release from locking by solute atom [8], critical plastic strain [16,17], critical dislocation density [6], and dislocation pile-up at the grain boundary [4] were closely associated with the sequential processes involved in determining the delay time for yielding. Thus, the theoretical results [8] based on Cottrell-Bilby's dislocation release mechanism [7] for dislocation locking by solute atoms (Equation (36)) are considered to be the starting for understanding plastic yielding.

The Applied Stress Rate Dependence of Yield Stress [17]

Strain rate caused by applied stress, $\dot{\gamma}_a$, is given by the summation of plastic strain rate of the specimen, $\dot{\gamma}_p$, and elastic strain rate including those of grips and rigidity of testing machine, $\dot{\gamma}_e$, as follows.

$$\dot{\gamma}_a = \dot{\gamma}_e + \dot{\gamma}_p \quad (38)$$

For the case of a sharp yielding point, $\dot{\gamma}_e \approx 0$ is satisfied at the yield point. Therefore, $\dot{\gamma}_a$ is given by Equation (39).

$$\dot{\gamma}_a = \dot{\gamma}_p \approx A^* \frac{1}{G} \dot{\tau}_a \quad (39)$$

where A^* is a proportional constant. By using Equations (11) and (12) for the number of dislocations emitted from a stressed source under constant stress rate condition and then substituting Equations (26) and (27) into Equation (39), σ_Y can be given by Equation (40) as a function of stress rate and material constants. Furthermore, $\tau_a = \tau_Y = \frac{\sigma_Y}{2}$ and $\dot{\tau} = \frac{\dot{\sigma}}{2}$ were also assumed.

$$\sigma_Y = 2\tau_0^* \left(\frac{m^{1.45}}{1.4} \frac{A^*}{\rho_0^* \Lambda b^2} \right)^{\frac{1}{(2m+1)}} \left(\frac{\tau_0^*}{G} \right)^{-\frac{2(m+1)}{(m+2)(2m+1)}} \left(\frac{\dot{\sigma} b}{2v_0 G} \right)^{\frac{(2m+3)}{(m+1)(2m+1)}} \quad (40)$$

Using Equations (7)~(10) and the following approximation for $m = 10$ –30 (steel) [16], Equation (42) was obtained.

$$\frac{2m+1}{(m+2)(2m+1)} \approx \frac{1}{m+2} \approx \frac{1}{m+1} \approx \frac{1}{m} \quad (41)$$

$$\sigma_Y = 2\tau_{00} \left\{ \frac{A^*}{1.4} \left(\frac{H_k}{4kT} \right)^{1.45} \frac{1}{\rho_0^* \Lambda b^2} \left(\frac{\tau_{00}}{G} \right)^{-1.9} \right\}^{\frac{2kT}{H_k}} \left(\frac{\dot{\sigma} b}{2A_1 G} \right)^{\frac{4kT}{H_k}} \quad (42)$$

As material constants, the following physically reasonable and almost equal values to those used for the analysis of the delay time of yielding, $\rho_0^* \Lambda = 1.75 \times 10^6 / \text{m}^2$ [16], $G = 79.38 \text{ GPa}$, $\tau_0^* = 176.0 \text{ MPa}$ [18], and $m = 10$ [16] were taken at room temperature, with $b = 3 \times 10^{-10} \text{ m}$ selected. Before yielding, since $\tau_a = G\gamma_a$ was almost satisfied, A^* was considered to be almost equal to one.

A comparison of Equation (42) with experimental data is shown in Figure 19 [17]. Equation (42) was found to agree well with experimental data [19]. Furthermore, this equation was in good agreement with that obtained based on dislocation dynamics theory that defined yielding as occurring when dislocation density takes some critical value [6] as follows.

$$\sigma_Y = 2\tau_{00} \left\{ \frac{1}{1.4} \left(\frac{H_k}{4kT} \right)^{1.45} \frac{\bar{N}_0}{\rho_0^* \Lambda} \right\}^{\frac{4kT}{H_k}} \left(\frac{G \dot{\sigma} b}{2A_1 \tau_{00}^2} \right)^{\frac{4kT}{H_k}} \quad (43)$$

This means that the γ_p criterion is identical to the \bar{N}_0 criterion. Furthermore, Equation (42) was found to be in good qualitative agreement with theoretical results [8] based on Cottrell-Bilby's dislocation release mechanism [7] for dislocation locking by solute atoms such as carbon or nitrogen,

as given by Equation (44). This means that the locking mechanism closely connects with the mechanism of dislocation group dynamics as the appropriate mechanism of locking.

$$\sigma_Y = \sigma_0 \left(\frac{t_0}{nkT} \frac{\dot{\sigma}}{\sigma_0} \right)^{nkT} \quad (44)$$

Furthermore, other results obtained based on a viscoplasticity model also showed the same type of relationship between strain rate and yield stress [5].

From the total results mentioned above, it can be seen the criteria of dislocation release from locking by the solute atom [8], critical strain rate [16,17], critical dislocation density [6], and viscoplasticity [5] are closely connected with the sequential processes of plastic yielding. Furthermore, the theoretical results [8] based on Cottrell-Bilby's dislocation release mechanism [7] from locking by solute atom (Equation (44)) are considered to be a starting process of yielding.

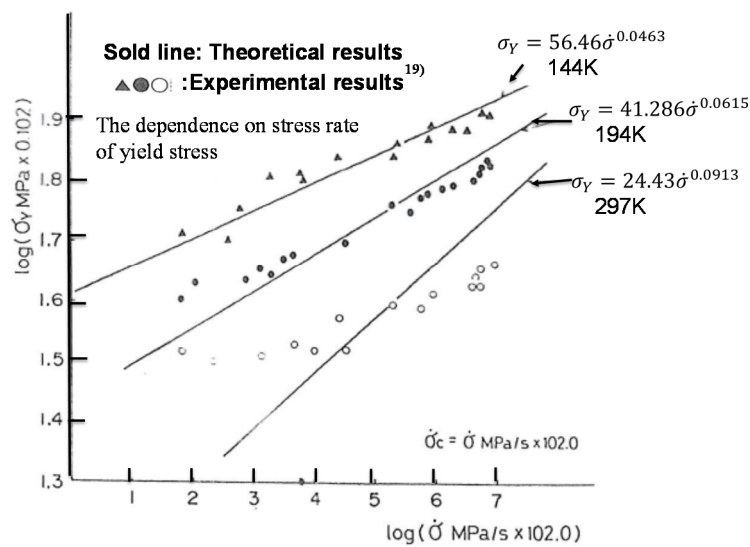


Figure 19. The relationship between upper yield point and constant stress rate based on dislocation dynamics. Solid lines represent the theoretical results [17]. Dotted lines represent experimental results [19].

T. Yokobori also found that the adoption of the friction stress, τ_i , of the motion of dislocation in Equation (40) was very effective in predicting experimental results with the current equation [20]. In this theory for Equation (40), the effect of τ_i on yield stress carries on to lead to Equation (45) [17].

$$\sigma_Y = \sigma_i + 2(\tau_0^* - \tau_i) \left(\frac{m^{1.45}}{1.4} \frac{A^*}{\rho_0^* \Lambda b^2} \right)^{\frac{1}{(2m+1)}} \left(\frac{\tau_0^* - \tau_i}{G} \right)^{-\frac{2(m+1)}{(m+2)(2m+1)}} \left(\frac{\dot{\sigma} b}{2v_0 G} \right)^{\frac{(2m+3)}{(m+1)(2m+1)}} \quad (45)$$

By adopting $\tau_i = 86.3$ MPa and using Equations (10a)–(10d), Equation (45) was found to well-predict experimental results [19], as shown in Figure 20 [17].

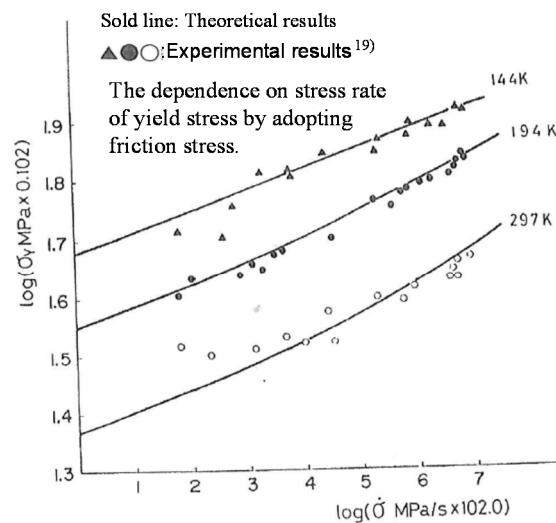


Figure 20. The relationship between the upper yield point and constant stress rate based on dislocation dynamics by accounting for the effect of friction stress of dislocation on upper yield point. Solid lines represent the theoretical results [17]. Dotted lines represent experimental results [19].

The relationship between τ_0^* used for analyses of Figures 17–20 and temperature T is shown in Figure 21. The results were in good agreement with the theoretical relationship given by Equation (10c), thus showing the validity of the method of analysis.

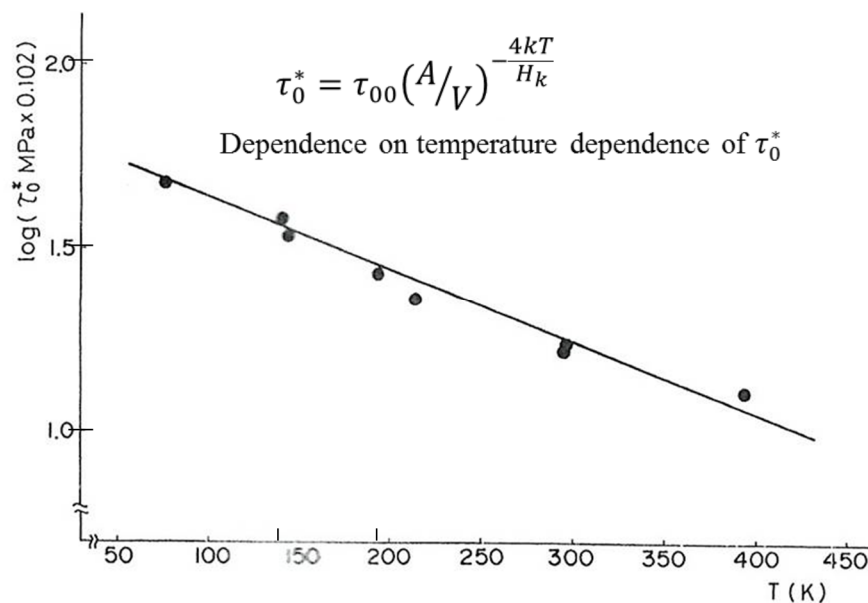


Figure 21. The relationship between τ_0^* , as used for this analysis, and temperature [17].

The Effect of Grain Size and Applied Strain Rate on Yield Stress Based on the Theory of Dislocation Piling Up [2,3,21]

The effect of grain size d on lower yield point was obtained by the following experimental relationship [22,23].

$$\sigma_{l,y} = \sigma_s + \kappa d^{-\frac{1}{2}} \quad (46)$$

where σ_s and κ are material constants that are positive values.

Furthermore, many detailed studies have been conducted on this relationship [24,25].

In this section, using the Equation (25), the calculated relationship between an applied stress, τ , required for a dynamic K_d to take on a critical value over a range in grain size d is shown by the solid line in Figure 22 [3], which is expressed by Equation (47).

$$\tau = \tau_1 + k \frac{1}{\sqrt{d}} \quad (47)$$

where τ_1 and k are constants. The dashed line is a static solution given by Equation (48).

$$\tau = \frac{K_s}{\sqrt{2\pi d}} \quad (48)$$

where K_s is static stress intensity factor.

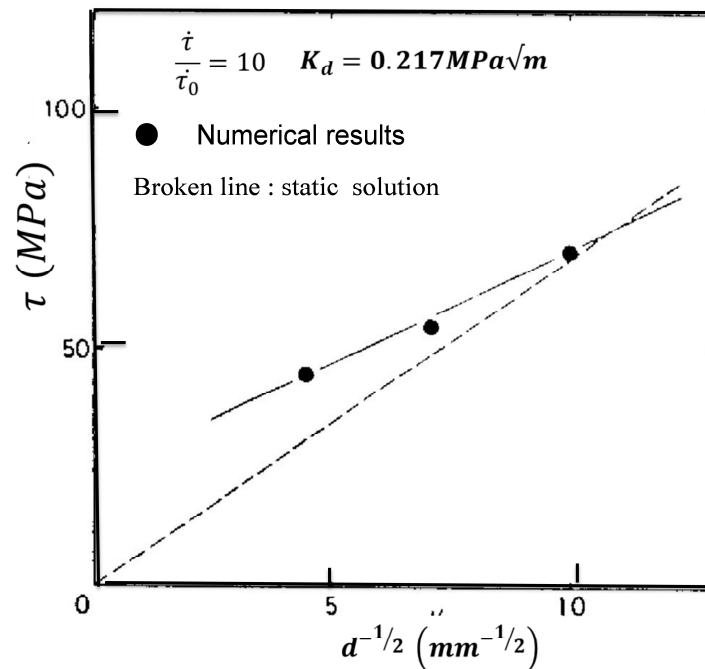


Figure 22. The relationship between yield strength and grain size, d . Solid line represents the results obtained by the analysis of dislocation dynamic pile, as shown by dotted data [3]. Dotted line represents the solution obtained by Equation (48).

It can be seen from Equation (47) that the dynamic yield strength also increases linearly with respect to the inverse square root of grain size. Furthermore, by comparing Equations (47) and (48), the yield stress corresponding to $d^{-\frac{1}{2}} \approx 0$ was found to be higher in dynamic yielding than that in the static case, and this characteristic is in good agreement with the experimental data [26].

Concerning the effect of applied strain rate on yield stress, the rate-determining process of yielding of iron and steel is considered to correspond to the dynamic piling up of emitted dislocations.

The dislocation pile-up at grain boundaries and yielding is considered to occur when K_d , given by Equation (25) to measure the local stress concentration, takes on a critical value. A comparison of results obtained by Equation (25) and experimental data [27] is shown in Figure 23 [3]. In Figure 23, the solid line represents the calculated relation between the applied stress σ required for K_d to take the critical value and applied strain rate that is in good agreement with experimental data [25]. In Figure 23, $\dot{\epsilon}$ is evaluated by the relationship of $\dot{\sigma} = 2\dot{\tau} = E\dot{\epsilon}$.

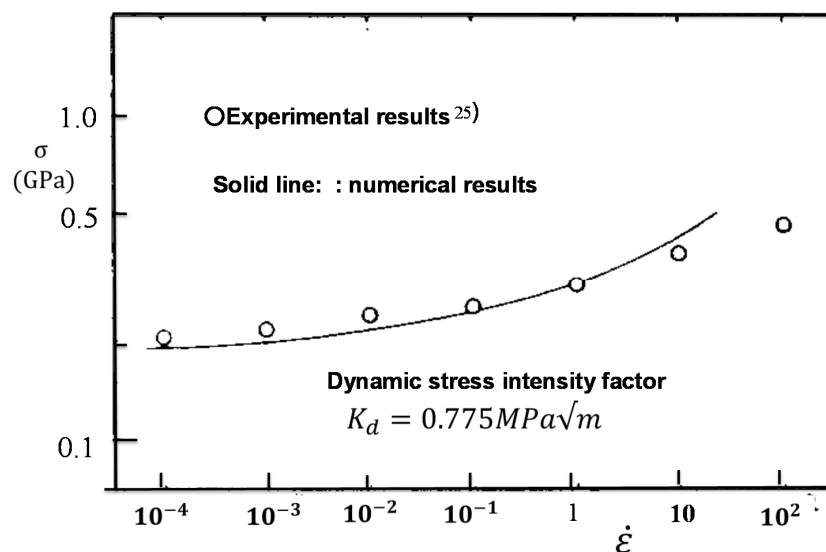


Figure 23. The relationship between strain rate and yield stress of mild steel. Dotted line represents the numerical results [3].

From the total results presented in Section 2.4, it can be seen that the various criteria associated with dislocation release from locking by solute atom [8], critical plastic strain [16,17], critical strain rate or stress rate, critical dislocation density [6], viscoplasticity [5], and dislocation pile-up at a grain boundary [3,4] characterized by local stress intensity factor [3] (K_d) all relate closely to the sequential processes of yielding. Furthermore, the theoretical results [8] based on Cottrell-Bilby's dislocation release mechanism [7] for dislocation locking by solute atoms is considered to be a starting process of yielding.

In addition to the relationship between yield stress and grain size, a theoretical relationship between the yield stress and temperature was derived based on dislocation mechanics [28,29]. The present description closely connects with the results given in Figures 18 and 20 [17].

2.5. Application to Problem of Creep

Previous descriptions of the creep rate have been dominated by the use of equations based on the properties of an isolated dislocation [30].

In this section, instead of using the velocity of an isolated equation, the results of dislocation group dynamics associated with emission from a stressed source under constant stress condition were adopted, and a creep rate dominated by the grouped dislocation mechanism was formulated.

The maximum radius of dislocation loop is given by Equations (49) and (50) [30].

$$L = \sqrt{\frac{2(\tau_a - \tau_i)}{Gb\rho_0^*}} \quad (49)$$

$$\Lambda^* = 3L \quad (50)$$

By using Equations (27)–(29) and Equation (32), the creep rate, $\dot{\gamma}$, can be given by Equation (51).

$$\dot{\gamma} = A_1(\tau_a - \tau_i)^\delta \quad (51)$$

In (51), τ_a in Equation (32) is replaced by $(\tau_a - \tau_i)$.

$$\delta = \frac{2m^2 + \frac{9}{2}m + 2}{m + 2} \quad (52)$$

$$A_1 = 3 \sqrt{\frac{2\rho_0^* b}{G}} \times 2.45m^{-0.86} \left(\frac{Gb}{v_0 t} \right)^{-\frac{m+1}{m+2}} \times (\tau_0^*)^{-\frac{m(2m+3)}{m+2}} v_0 \quad (53)$$

Furthermore, when using Equations (7)–(10), the creep rate, $\dot{\gamma}$ is given by Equation (54) as an equation of a thermally-activated process.

$$\dot{\gamma} = C_1 (\tau_a - \tau_i)^{\alpha_1} \exp\left(-\frac{Q_1}{kT}\right) \quad (54)$$

where

$$C_1 = 3 \sqrt{\frac{2\rho_0^* b}{G}} \times 2.45m^{-0.86} A_1 \left(\frac{Gb}{A_1 t} \right)^{-\frac{m+1}{m+2}} \times (G)^{-\frac{m(2m+3)}{m+2}} \quad (55)$$

$$\alpha_1 = \frac{2m^2 + \frac{9}{2}m + 2}{m + 2} \quad (56)$$

$$Q_1 = \frac{2m + 3}{4(m + 2)} H_k \ln \frac{\tau_{00}}{G} \quad (57)$$

Equation (54) is in good agreement with a pioneering experimental equation given for the creep rate [31]. Since $m = 1$ is valid for Zn [13], α_1 is 2.83 in Equation (54), which is in good agreement with experimental data [31]. Thus, Equation (54) is a theoretical equation of a creep rate dominated by a dislocation mechanism that incorporates the effect of dislocation dynamics corresponding to dislocations being emitted from a stressed source.

2.6. Application to the Problem of Fatigue Crack Growth [32,33]

The fatigue crack growth rate da/dN for a crack blunting and re-sharpening model [34] is approximately equal to $\frac{1}{2}U$, as shown in Figure 24 [32,33] and given by Equation (58).

$$\frac{da}{dN} \cong \frac{1}{2}U = nb \quad (58)$$

where U is the crack opening displacement caused by dislocation emission from a crack tip and is equal to $2nb$, b is the Burgers vector, and n is the number of dislocations emitted from a crack tip, as shown in Figure 24.

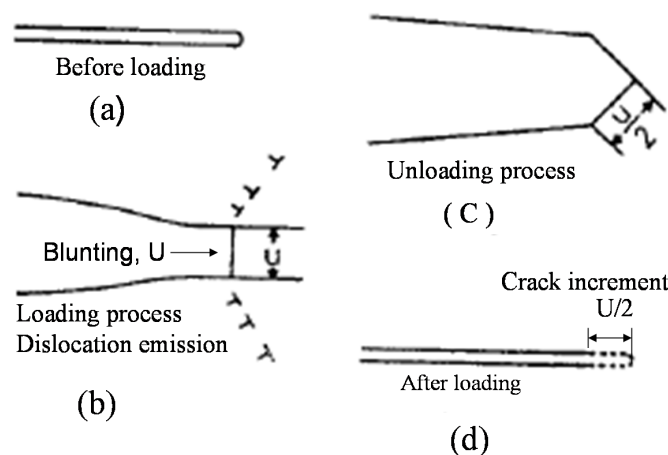


Figure 24. Blunting and re-sharpening model of fatigue crack growth. (a) Before fatigue load cycle, (b) maximum load of the fatigue cycle, (c) unloading process of the fatigue cycle, and (d) complete unloading of the fatigue cycle. The crack increment of $U/2$ (crack opening displacement/2) is caused by dislocation emission.

Substituting Equations (6)–(10) into Equation (58) allows for the fatigue crack growth rate to be expressed in terms of an apparent single thermal activated process that is given by Equations (59a) and (59b) [32,33].

$$\frac{da}{dN} = b\gamma(m) \left(\frac{4fb}{v_0} \right)^{-\frac{m+1}{m+2}} \left(\frac{\tau_0^*}{G} \right)^{-\frac{m(m+1)}{m+2}} \left(\frac{\Delta K_1}{\sqrt{\varepsilon}G} \right)^{\frac{(m+1)^2}{m+2}} \quad (59a)$$

$$= bA^* (4f)^{-\frac{m+1}{m+2}} \left(\frac{\Delta K_1}{\sqrt{\varepsilon}G} \right)^{\frac{m+1}{m+2}} \exp \left\{ -\frac{\left(\frac{m+1}{m+2} \right) H_k \ln \left(\frac{\tau_{00} \sqrt{\varepsilon}}{\Delta K_1} \right)}{4kT} \right\} \quad (59b)$$

where ΔK_1 is the stress intensity factor, f is load frequency, and ε is the local distance from a crack tip, e.g., 1.5×10^{-4} mm. Local stress around a crack tip is characterized given by $\tau_a = \frac{\Delta K_1}{\sqrt{\varepsilon}}$.

$$A^* = 1.4m^{-1.45} \left(\frac{b}{A_1} \right)^{-\frac{m+1}{m+2}}$$

Equations (59a) and (59b) can be written as Equation (60).

$$\ln \frac{da}{dN} = \ln \left(\frac{A_2}{f^\lambda} \right) + b_0 \ln \Delta K_1 - \frac{U_2 - a_2 \ln \Delta K_1}{kT} \quad (60)$$

where A_2 , a_2 , and b_0 are material constants.

Equations (59a) and (59b) can be expressed as:

$$\frac{da}{dN} = B_1 \Delta K_1^\delta \quad (61)$$

Which is the well-known experimental Equation by Paris [35], where $\delta = \frac{(m+1)^2}{(m+2)}$.

The experimental relationship between $\ln \left(\frac{da}{dN} \right)$ and $1/T$ is shown in Figure 25 [36] with the parameter of stress intensity factor amplitude, ΔK_1 .

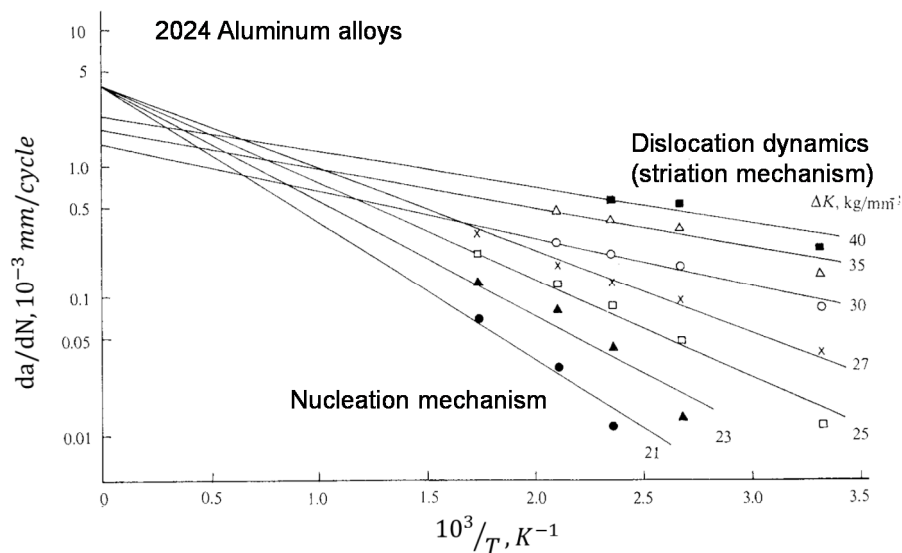


Figure 25. The thermally activated relationship between da/dN (fatigue crack growth rate) and the inverse value of absolute temperature, $\frac{10^3}{T}$, for 2024 aluminum alloys [36].

Figure 25 shows that these relationship were found to hold for the thermally activated process in the range of higher values of ΔK_1 , in the intercept values of the straight line of $\ln \left(\frac{da}{dN} \right)$ with

the coordinate axis at $1/T = 0$ were an approximately linear function of $\ln(\Delta K_1)$, which is in good agreement with Equation (60), that is, $\ln\left(\frac{A_2}{f^\lambda}\right) + b_0 \ln \Delta K_1$ [32,33].

On the other hand, in the range of lower values of ΔK_1 , the intercept value is constant with stress intensity factor amplitude, ΔK_1 , which is in good agreement with the model based on micro crack nucleation at the crack tip [37], given by Equation (62).

$$\ln \frac{da}{dN} = \ln\left(\frac{A_1}{f}\right) - \frac{U_1 - a_1 \ln \Delta K_1}{kT} \quad (62)$$

where A_1 and a_1 are material constants. The same experimental tendencies were also found in the relationship between $\ln\left(\frac{da}{dN}\right)$ and $1/T$ for stainless steel [38]. Furthermore, Equation (60) was found to be in good agreement with the experimental relationship between $\ln\left(\frac{da}{dN}\right)$ and $1/T$ for high strength steel at low temperatures [39].

Experimental data showed that da/dN is proportional to $f^{-\lambda}$, and λ experimentally takes values from 0.1 to 0.2 for steel [40] and 0.1 to 0.5 for aluminum alloys [41].

For the case of an elastic–plastic crack, the local stress around a crack tip is written by Equation (63) [42].

$$\sigma_l = f(\beta) \sigma_{cy} \left(\frac{\Delta K_1}{\sigma_{cy} \sqrt{\varepsilon}} \right)^{\frac{2\beta}{1+\beta}} \quad (63)$$

where σ_{cy} , β , and $f(\beta)$ are the initial yield stress in cyclic straining, cyclic strain hardening exponent, and some function of β , respectively.

Therefore, by comparing Equations (63), (61), and (59a), da/dN can be given by the following equation [33].

$$\frac{da}{dN} \propto \left(\frac{\Delta K_1}{\sqrt{2\varepsilon} \sigma_{cy}} \right)^{\frac{2\beta}{1+\beta} \frac{(m+1)^2}{m+2}} \quad (64)$$

Furthermore, for the effect of multiple slip lines and strain hardening under cyclic loading, Equation (64) can be rewritten as Equation (65) [33].

$$\frac{da}{dN} \propto \left(\frac{\Delta K_1}{\sqrt{2\varepsilon} \sigma_{cy}} \right)^{\frac{2\beta}{1+\beta} \frac{(m+1)^2}{m+2} + \frac{1}{1+\beta}} \quad (65)$$

For $\beta = 0.08–0.3$ and $m = 4–10$, which are reasonable values for steel and aluminum alloys, the power exponent becomes $\delta = 2.0–5.0$, which are also experimentally reasonable values.

In Equation (61), the following equation can be seen from Equation (59a).

$$B_1 = B / (\sqrt{\varepsilon} G)^\delta \quad (66)$$

From Equation (66), Equation (67) can be obtained [43].

$$\ln B_1 = \ln B - \delta \cdot \ln(\sqrt{\varepsilon} G) \quad (67)$$

Equation (67) was found to be in good agreement with the experimental relationship [43] between B_1 and δ , as shown in Figure 26 [43].

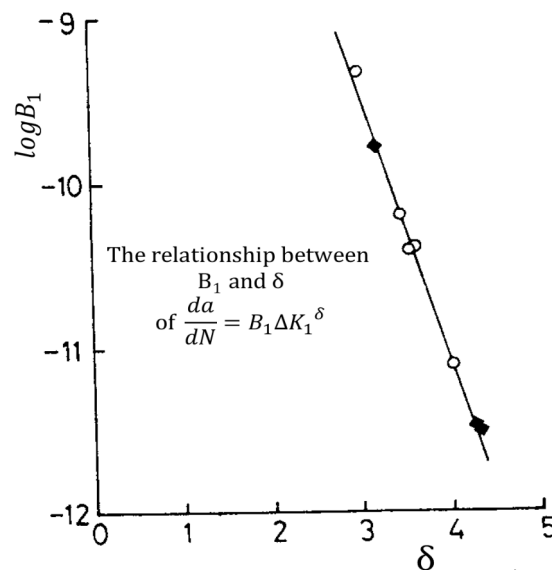


Figure 26. The experimental relationship between B_1 and δ in Equation (67) [43].

3. Concluding Remarks and Future Problem

Analyses of discrete dislocation dynamics and emission from a stressed source were conducted. The results obtained by these analyses enabled us to link various dynamical effects, such as load frequency and temperature, on the yield stress, dislocation creep rate, and fatigue crack growth rate with the experimental results of macroscopic phenomenon and to also link them with theoretical results obtained by the concept of static, continuously distributed infinitesimal dislocations for the equilibrium state under low strain or stress rate conditions.

This will be useful as a holistic research approach relating to the time scale—e.g., ranging from results under high strain rate condition to those under static or low strain rate conditions—and the space scale—e.g., ranging from meso-scale and macro-scale mechanics—that is, from the scale of dislocation groups dynamics to fracture mechanics.

To establish a perfect link of mesoscopic mechanics with macro mechanics and for practical applications to engineering structures, further nonlinear interactive treatments will be necessary, e.g., effects of vacancy diffusion, different multiaxial stress in structures, and different scales of grain boundary influences on dislocation group dynamics. For these study fields, the establishment of interdisciplinary science between material science and structural engineering coupled with computational mechanics is needed as one of future research problems involving the strength of materials.

Detailed research on the effects of grain size and temperature on the yield stress has been systematically conducted, and many innovative results have been obtained [28,29].

The proposed research approach mentioned in this article will enable us to link mesoscopic mechanical factors with macro-scale engineering results [28,29].

Concerning the problems of nano-scale fracturing and plasticity, many studies based on the method of 3D discrete dislocation dynamics have already been successfully conducted [1]. These studies would appear to directly connect with physical properties of dislocations and nano-scale fracturing behaviors. The present results should lead to a wider establishment of fracture prediction in the full range from the nano-scale to macro-scale.

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