



Article An Iterative Modified Adaptive Chirp Mode Decomposition Method and Its Application on Fault Diagnosis of Wind Turbine Bearings

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Abstract: Wind turbine bearings usually work with strong background noise, making the faulty properties difficult to extract and detect. To accurately diagnose the faults of rolling bearings in wind turbines, an iterative modified adaptive chirp mode decomposition (IMACMD) method is proposed in this paper. Firstly, an envelope interpolation method is employed to preliminarily determine the iterative mode number and guide the potentially initial frequency selection. Secondly, the upper limits of the iterative mode number and the initial frequency are further determined through correlation analysis. During the iteration process, the optimal weight factor of the reconstructive input signal, which is the residual signal of the previous iterative decomposition, is determined according to the new designed ensemble *L*-Kurtosis index. Experimental and engineering signals are used to validate the proposed IMACMD method. Comparisons with the conventional methods demonstrate the superiority of this proposed method. It is shown that this method can not only identify the weak features for single faults but also separate the multiple features for compound faults.

Keywords: fault diagnosis; parameter estimation; iterative modified adaptive chirp mode decomposition (IMACMD); envelope interpolation; ensemble *L*-Kurtosis

1. Introduction

Rolling bearings are key supporting components and have been widely used in a lot of industrial applications, such as high-speed trains, helicopter drive trains, wind turbines, etc. The running state of bearings greatly affects the safety of the system [1], since their unexpected failure leads to machinery breakdown or even catastrophic accidents [2]. Due to the reasons of heavy load, long term, and harsh operation environment in modern industry, the critical bearings in rotating machines have high potential of being damaged. Therefore, it is of great significance to accurately diagnose the bearing faults [3]. The damage may be concentrated on either one part (single fault) or multiple parts (compound fault). Compared to a single fault, compound faults are more harmful to the bearings [4] since they have mutual-coupling and cross-influence effects on the neighboring components [5]. Usually the diagnosis method for a single fault is difficult to use for compound faults [6,7]. Therefore, new diagnosis methods that are not only effective for both single and compound faults, but also qualified to accurately identify the faulty types and locations, are urgently requested.

At present, the vibration-based analysis method is the most convenient and widely employed bearing fault diagnosis technology, though the incipient failures cause some changes to the operating condition information in the collected vibration signals. According to the dynamics and fault mechanism, when a local injury appears on the rolling bearing, the high-frequency structural resonance will also be reflected in the bearing system [8]. The fault characteristic is generally regarded as contained in a periodic impact signal [9]. In



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Copyright: © 2022 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC BY) license (https:// creativecommons.org/licenses/by/ 4.0/). theory, when the periodic impacts match the theoretical properties of the bearing, the fault type and location can be determined. In practice, due to the influences of many factors such as the shaft vibration, gear meshing, electromagnetic interferences, and external environmental issues, there may be harmonics, accidental pulses, and noises in the vibration signal [10]. Consequently, a weak fault signal is difficult to identify [11]. More seriously, when the compound faults occur, the coupling effect between two different faults may make the fault shocks overlap and even compensate each other, making the fault shocks become weaker and the compound fault diagnosis much more challenging [12].

In the past few decades, incipient fault diagnosis on rolling bearings has attracted the extensive attention of many scholars, who have carried out a series of studies for guiding industrial applications [13–15]. The popular and powerful signal-processing algorithms in bearing fault diagnosis mainly include spectral kurtosis (SK), deconvolution algorithms, and decomposition methods. As evolved versions of SK, SKRgram [16], Infogram [17], Autogram [18], and Accugram [19] are still restricted by the best frequency band selection [20] and the fixed spectral segmentation strategy, which may lead to the loss of some key faulty information [21]. The deconvolution algorithm, such as the minimum entropy deconvolution (MED) [22], maximum correlated kurtosis deconvolution (MCKD) [23], multipoint optimal minimum entropy deconvolution adjusted (MOEDA) [24], and cyclostationary blind deconvolution (CYCBD) [25], iteratively updates the filter to make the target evaluation index of the output signal reach the optimal value, so that the fault-affected signal can be recovered [26]. The performance of deconvolution algorithm greatly depends on the selection of the objective function and the filter length, which limit the practical application of the deconvolution methods [27]. Both the deconvolution and SK methods mainly focus on the extraction of the dominant fault features, whereas the secondary fault feature may be ignored while processing the vibration signals of the compound faults [28]. The signal decomposition methods, such as empirical mode decomposition (EMD) [29], local mean decomposition (LMD) [30], singular spectrum decomposition (SSD) [31], and variational mode decomposition (VMD) [32], are designed to decompose the signals into mode components of different frequency bands and separate the fault characteristic signals from the interference signals. As a very suitable technique for bearing fault detection [33], the main disadvantages of signal decomposition methods are as follows: (1) The original vibration signal will be decomposed into many sub-components, making the processing results very complicated, and (2) many parameters need to be well preset, since the satisfactory analysis results depend on the accurate setting of each parameter [34].

As a novel non-stationary and nonlinear signal analysis algorithm, adaptive chirp mode decomposition (ACMD) is developed based on the greedy search algorithm [35] and is able to extract the particular mode containing rich characteristic information without other redundant modes. The mode obtained by ACMD is a band-limited signal in which the primary energy is concentrated around a center frequency [36]. Compared with the aforementioned analysis methods, ACMD requests fewer preset parameters. Different from EMD, LMD, SSD, and VMD algorithms, which separate all the modes simultaneously, ACMD only obtains a particular mode containing rich characteristic information at a time to improve the time-frequency resolution. ACMD has been applied to the fault diagnosis on rolling bearings by scholars [37,38]. The authors have also studied this method for rolling bearing fault diagnosis, and more details can be found in [39].

However, most of the previous work focuses on single faults, and few of them have paid attention to compound faults, which can also take place in rolling bearings. To promote existing methods such as ACMD being effective in feature extraction for both single and compound faults, the following three issues need to be primarily considered: (1) The instantaneous frequency initialized in ACMD should be determined adaptively without prior knowledge, (2) proper modifications need be carried out to improve current methods such as ACMD to make them effective in detecting the all potential components for the weak and compound failures from the original signals, and (3) as the key parameter of ACMD, given that the weight factor directly affects the bandwidth of the estimated mode in ACMD, such key parameters need to be set reasonably. To resolve the aforementioned three issues, in this paper we propose an iterative modified adaptive chirp mode decomposition (IMACMD) method, which can further expand its application field from single faults to compound faults. What's more, this method is able to detect all the sensitive components from the original signal.

Our proposed method primarily includes three steps. Firstly, an envelope interpolation method is adopted to preliminarily determine the number of decomposition modes and to guide the instantaneous ACMD frequency of each mode. Then, the modal number and the instantaneous frequencies can be obtained by deleting the strongly correlated modes.

Secondly, considering the interaction of the compound fault components, the input signal decomposed by the next iteration is reconstructed after filtering the specific frequencies through ACMD. In each iteration, the input signal of the first iteration is the original signal, whereas the subsequent input signal is the remaining one after removing the filtering signal.

Finally, based on the advantages of the time-domain *L*-Kurtosis and the frequencydomain *L*-Kurtosis, the ensemble *L*-Kurtosis (*ELK*) is designed as a new index. Under the guidance of the modal number and the instantaneous frequencies in the aforementioned steps, the new index, which considers both the cyclostationary process and the impulsiveness, is used to select the optimal weight factor (α) for each IMACMD decomposition mode.

Through these improvements, the above-mentioned issues are attempted to be solved. The feasibility of the new IMACMD method are verified by simulating, experimental, and engineering signals. At the same time, this paper also tries to compare SSD and SK, and discusses the superiority of the proposed method. Using the IMACMD method, more ideal fault diagnosis results of rolling bearings are expected to be obtained.

The remainder of this paper Is organized as follows. Section 2 describes the basic principles of ACMD, analyzes the effects of its key parameters, and proposes and discusses the signal evaluation index *ELK*. In Section 3, the particular implementation steps of the proposed IMACMD method, including the localization strategies for the number of iterations, the instantaneous frequency, and the weight factor of ACMD, are presented in detail. Subsequently, the effectiveness and the superiority of IMACMD are validated by the experimental signals and engineering signals in Sections 4 and 5, respectively. Finally, conclusions are summarized in Section 6.

2. Theoretical Description and Characteristics Study on ACMD

2.1. Basic Theory of ACMD

Extended from VMD, ACMD is a novel algorithm for non-stationary signal decomposition and has obvious advantages in the processing of multi-component strong frequency modulation signals and excellent time-frequency resolutions [40]. Since ACMD is sensitive to the impact responses, it has been treated as a good choice to process the vibration signals for the bearing fault diagnosis [41].

The main parts of ACMD consist of a recursive mode extraction framework, an adaptive bandwidth updating rule, and a frequency initialization scheme based on Hilbert transform. For a nonstationary signal x(t), ACMD solves the optimization problem as follows:

$$\min_{a_m(t), b_m(t), f_m(t)} \left\{ \left\| p_m''(t) \right\|_2^2 + \left\| q_m''(t) \right\|_2^2 + \alpha \| x(t) - x_m(t) \|_2^2 \right\}$$
(1)

with

$$x_m(t) = p_m(t)\cos\left(2\pi\int_0^t \widetilde{f}_m(t)dt\right) + q_m(t)\sin\left(2\pi\int_0^t \widetilde{f}_m(t)dt\right)$$
(2)

where $p_m(t)$ and $q_m(t)$ are both the de-chirped signals, α denotes the weight factor, $f_m(t)$ represents the instantaneous frequency, $||x(t) - x_m(t)||_2^2$ is the residue energy after the current estimated component is removed, and the square of the second derivative is employed to measure the bandwidth of the target mode.

The signal x(t) is discretized to a discrete signal whose time series is $t = t_0, \dots, t_{N-1}$, and the number of the sampling points is N; the discrete version of Equation (1) can be expressed as:

$$\min_{u_m, f_m} \left\{ \| \boldsymbol{\Theta} \mathbf{u}_m \|_2^2 + \alpha \| \mathbf{x} - \mathbf{G}_m \mathbf{u}_m \|_2^2 \right\}$$
(3)

where $\boldsymbol{\Theta} = \begin{bmatrix} \boldsymbol{\Omega} \\ \boldsymbol{\Omega} \end{bmatrix}$, and $\boldsymbol{\Omega}$ is a second-order difference matrix. $\mathbf{u}_m = \begin{bmatrix} \mathbf{p}_m^T, \mathbf{q}_m^T \end{bmatrix}^T, \mathbf{p}_m = [p_m(t_0), \cdots, p_m(t_{N-1})]^T, \mathbf{q}_m = [q_m(t_0), \cdots, q_m(t_{N-1})]^T$, and $\mathbf{x}_m = [x_m(t_0), \cdots, x_m(t_{N-1})]^T$. The kernel matrix \mathbf{G}_m is related to the demodulation frequency $\tilde{f}_m(t)$ and can be written as:

$$\begin{cases}
\mathbf{G}_{m} = [\mathbf{C}_{m}, \mathbf{D}_{m}] \\
\mathbf{C}_{m} = diag[\cos(\varphi(t_{0})), \cdots, \cos(\varphi(t_{N-1}))] \\
\mathbf{D}_{m} = diag[\sin(\varphi(t_{0})), \cdots, \sin(\varphi(t_{N-1}))] \\
\varphi_{m}(t) = 2\pi \int_{0}^{t} f_{m}(t) dt
\end{cases}$$
(4)

x(t) is decomposed by minimizing the demodulation signal bandwidth, and the signal components are estimated one by one to achieve the accurate estimation on the multi-component chirp signals.

2.2. Ensemble L-Kurtosis Indicator

When rotating machineries are damaged, periodic pulse signals will appear. Such pulse signals are the most important basis for fault detection. In a weak noise environment, the time-domain features, including the pulses and periodicities, are easy to recognize. However, when the fault signal is fulfilled by noises, the fault characteristics are easy to bury in the time-domain waves but can be found in the spectrum. Although the bearing fault information exists in the full spectrum, most of the energy is concentrated near the natural frequencies. Ideally, a composite signal can be decomposed into a series of independent components, several of which contain the fault information and can be visually displayed in the envelope spectrum.

Appropriate signal characteristic evaluation criteria are of great significance for mechanical fault diagnosis. Kurtosis (ku) [42] is considered one of the sparsity measurements, since it can detect the impulsiveness of the signal. Similar to the traditional ku indicator, *L*-Kurtosis is defined as an alternative indicator to evaluate the variable distribution characteristics by extending the traditional moment principle [43], and it has better antiinterference and robustness properties than Kurtosis [44].

Assume that $X = [X_1, X_2, \dots, X_n]$ is an independent sample from the cumulative distribution P(x), and $X_{1:n} \le X_{2:n} \le \dots \le X_{n:n}$ is a random variable drawn from X. The *r*th L-moment μ_r of the independent sample X is described as:

$$\mu_r = \frac{1}{r} \sum_{k=0}^{r-1} \left(-1\right)^k \binom{r-1}{k} E(X_{r-k:r}), r = 1, 2, \cdots$$
(5)

where the expectation $E(X_{r-k:r})$ is expressed as:

$$E(X_{j:r}) = \frac{r!}{(j-1)!(r-j)!} \int_0^1 x [P(x)]^{j-1} [1-P(x)]^{r-j} dP(x)$$
(6)

The 2nd and the 4th L-moments can be calculated as:

$$u_2 = \int_0^1 x(2P(x) - 1)dP(x) \tag{7}$$

$$\mu_4 = \int_0^1 x(20P^3(x) - 30P^2(x) + 12P(x) - 1)dP(x)$$
(8)

Then the definition of *L*-kurtosis is expressed as:

$$L - \text{Kurtosis} = \mu_4 / \mu_2 \tag{9}$$

However, *L*-Kurtosis pays more attention to the strength of the impacts rather than the regularity of the impacts. The spectrum of *L*-Kurtosis, which can be used to evaluate the cyclostationary process, is used in the fault diagnosis. Therefore, a new index is constructed by the virtue of *L*-Kurtosis and the spectrum of the *L*-Kurtosis. It is defined as

$$ELK = Lku \cdot FLku \tag{10}$$

where *ELK* represents the ensemble *L*-Kurtosis, *Lku* is the time *L*-Kurtosis, and *FLku* is the spectrum of the *L*-Kurtosis.

To investigate the performance of the above indexes, a simulating signal was constructed based on the vibration model in Section 2.3. As shown in Figure 1, the simulating signal consists of four components. Besides the fault feature and noise, random impulses from the electromagnetic interference frequency and the harmonic components of the shaft rotating appeared in the simulated signal. The detailed parameters are presented in Table 1. The normalized results of the four indexes used to calculate each simulated component are shown in Figure 2. The comparison shows that *ku* and *Lku* were most sensitive to the random impulses, rather than the fault impulses and harmonic components. On the contrary, *Flku* was most sensitive to the harmonic components. Obviously, *ELK* focused more on the cyclic impact characteristics than *ku*, *Lku*, and *FLku*. In addition, compared with *Lku* and *FLku*, *ELK* can employed better noise reduction. Considered a satisfactory indicator, *ELK* could accurately quantify the abundance of the periodic shocks. Consequently, it was able to capture the fault information and the harmonic information to the maximum extent.



Figure 1. Components of the simulated signal: (**a**) fault feature signal; (**b**) random impulses; (**c**) harmonic components; (**d**) white noise.

Parameter	Parameter Meaning			
	Fault feature signal			
I	Number of cycle impacts 165			
Α	Amplitude of cycle impacts	1.1		
f_n	Resonance frequency of cycle impacts	5500 Hz		
T_a	Period of cycle impacts 0.0083 s			
γ	Specifies the slippage characteristic	random value in 1–2% T_a		
	Random impulses			
J	Number of random impulses	3		
B_1, B_2, B_3	Amplitude of <i>j</i> th random impulse 2, 3, 4			
f_v	Resonance frequency of random impulses 4000 Hz			
T_1, T_2, T_3	Occurrence time of <i>j</i> th random impulse	0.08 s, 0.20 s, 0.25 s		
	Harmonic components			
K	Number of harmonic components	8		
<i>C</i> ₁ – <i>C</i> ₈	Amplitude of <i>k</i> th harmonic component 0.7, 0.8, 0.6, 0.5 0.3, 0.5, 0.4, 0.3			
f1-f8	Resonance frequency of <i>k</i> th harmonic	300, 400, 500, 600,		
	component	2850, 3000, 3150, 3300		
$arphi_k$	Phase of <i>k</i> th harmonic component	$\pi/2$		

Table 1. Setting of the simulated signal.



Figure 2. Comparison of the four indexes.

2.3. Research on the Influence of ACMD Parameters

2.3.1. Bearing Fault Simulating Signal

As two significant parameters of ACMD, the optimization of the instantaneous frequency f_c and the weight factor α has been studied by many scholars. However, which of these two parameters is more important has been rarely discussed in the previous works. To qualitatively study this topic, the observation signal of a single bearing fault is constructed as

$$x(t) = \underbrace{\sum_{i=1}^{I} A_{i} \sin(2\pi f_{n}t) s_{i}(t - iT_{a} - \gamma_{i})}_{fault \ feature} + \underbrace{\sum_{j=1}^{I} B_{j} s_{j}(t - jT_{j})}_{random \ impacts} + \underbrace{\sum_{k=1}^{K} C_{k} \sin(2\pi f_{k}t + \phi_{k})}_{harmonic \ components} + \underbrace{n(t)}_{noise}$$

$$(11)$$

The simulated fault signal consists of the fault feature signal, the random impulses, the harmonic components, and the noise. In this simulating verification, $s_m(t)$ represents the impulse response function of the rotating machinery system and can be expressed as follows:

$$s_m(t) = \exp(-\beta_m t)\cos(2\pi f_m t + \phi_m) \tag{12}$$

where β_m , f_m , and φ_m denote the coefficient of the resonance damping, the resonance frequency, and the phase, respectively.

The parameter setting of the simulated signal is shown in Table 1. The sampling frequency $f_s = 12$ kHz and the sample length $N_s = 8192$. The number of random impacts caused by the occasional shocks and the electromagnetic interferences on the machine was set as 3. The Gaussian distributed white noise n(t) = 0.5randn $(1, N_s)$ was added to the simulating signal. In the meantime, an inner race weak defect signal was generated, with a fault characteristic frequency $f_{in} = 120$ Hz. The components of the simulating signal are illustrated in Figure 1a,d. The resonant frequency band of the simulated inner race fault was set as 5500 Hz. On account of some uncertain interference components with high energy, the energy of this resonant frequency band was very weak. From the waveform illustrated in Figure 3a, the time intervals between two adjacent shocks did not match the period of the cycle impacts. Figure 3b shows the spectrum of the composite simulating signal; it can be seen that the main energy was not concentrated at 5500 HZ. Furthermore, the most important fault information at 120 Hz was invisible in the envelope spectrum depicted in Figure 3c.



Figure 3. Simulated signal: (a) waveform; (b) frequency spectrum; (c) envelope spectrum.

2.3.2. Study on Decomposition Characteristics of ACMD with Different Parameters

Based on the variable-control method, the simulated signal x(t) was used to study the feature extraction effect of ACMD with different f_c and α .

Firstly, the influence of f_c on the ACMD filtering effect was analyzed. α was fixed to 0.1, whereas f_c was set to 500 Hz, 1500 Hz, 2500 Hz, 3500 Hz, 4500 Hz, and 5500 Hz. The envelope spectra of ACMD filtering signals with different f_c values are shown in Figure 4. When f_c was set to the resonant frequency ($f_n = 5500$ Hz) of the fault signal, the fundamental frequency ($f_{in} = 120$ Hz) and the double frequency ($2f_{in} = 240$ Hz) of the characteristic component could be effectively extracted. In the other five groups of ACMD filtering signals, the harmonic and the random impact information could be recognized, but there was little identifiable fault feature information. This indicates that more useful characteristics can be captured if f_c is properly configured.

Subsequently, the effect of α on the ACMD algorithm was studied under the optimal instantaneous frequency condition, with $f_c = 5500$ Hz. The envelope spectrum of each filtering signal of ACMD is shown in Figure 5a, with α set to 10^{-1} , 10^{-2} , 10^{-3} , 10^{-4} , and 10^{-5} , respectively. According to the five envelope spectra, the following conclusions can be drawn: (1) The third spectrum ($\alpha = 10^{-3}$) had the highest fundamental frequency amplitude (fault characteristic), whereas the first and the fifth ones had the lowest amplitudes. (2) The interference components were obviously eliminated with the decrease in α , but the fault information was first enhanced and then attenuated. (3) When $\alpha = 10^{-3}$, the fault information extraction was obviously the most powerful, and $f_{in} \sim 4f_{in}$ were clearly visible. (4) After f_c was set correctly, the further reasonable selection of α made the characteristic information

much more clear. Figure 5b displays the histogram of the *ELK* indicators corresponding to different α values, which further validates the effectiveness of this indicator in evaluating the richness of the periodic impact components. Thus, *ELK* can be applied as the guidance indicator to optimize the parameter α for ACMD.



Figure 4. The envelope spectrums of filtering signals by ACMD with different f_c .



Figure 5. Filtering signals by ACMD with different α : (a) envelope spectrums; (b) *ELK* value.

By the comparative analysis results in Figures 4 and 5, it can be concluded that the weight factor had less impact than the instantaneous frequency on the execution effect of the ACMD algorithm. Therefore, the instantaneous frequency parameter should be accurately confirmed first, or else the fault characteristic information cannot be correctly captured. In order to effectively separate the optimal target mode from others, an innovative iterative ACMD fault diagnosis method based on the envelope interpolation and the correlation analysis is proposed; more details are presented in the subsequent sections.

3. Proposed IMACMD Method

3.1. Overview of the Novel Iteration Fault Diagnostic Strategy

Bearing faults stimulate the resonance of the system. Theoretically, the fault characteristic signal is a narrow band signal with a specific excitation resonance frequency as its center frequency. However, in practice, the fault features are mixed with a lot of interference. As a direct fault feature extraction method, ACMD has some limitations. It is difficult to extract the fault information, especially for compound faults. It is also difficult to estimate the instantaneous frequency and the weight factor of ACMD accurately. Hence, in this paper, a novel iterative modified adaptive chirp mode decomposition (IMACMD) method is proposed to maximize the advantages of the ACMD algorithm. Firstly, the maximum iteration number (K) and the instantaneous frequency (fc) of each iteration are determined by envelope interpolation and similarity evaluation. Secondly, according to the input signal of each iteration, the weight factor (α) of the ACMD algorithm in each iteration is optimized adaptively and reasonably. Finally, the envelope spectra of the iteration filter signal are compared with the theoretical fault characteristic frequency to judge the fault types. The proposed IMACMD method is explained by the flowchart shown in Figure 6.



Figure 6. Implementation steps of the proposed IMACMD.

3.2. Determination of Maximum Iteration Number (K) and Instantaneous Frequencies (f_c) of ACMD

3.2.1. Initialization

First, an envelope interpolation method is employed to preliminarily guide the potential instantaneous frequencies. It is applied to get the improved spectrum of the original signal by reducing the local maximum values and estimating the total f_c close to the reasonable value. The process is illustrated in the yellow box in Figure 6. The detailed steps of this module can be summarized as follows:

(1) Load the collected vibration signal *x*.

(2) Initialize the iteration number n = 1.

(3) Obtain the frequency signal y_{n-1} of the original signal *x*.

(4) The number of the local maximum values of y_{n-1} is calculated and written as k_{n-1} .

(5) y_{n-1} is processed by the interpolation envelop method, and the new frequency enveloped signal is denoted as y_n .

(6) Calculate the number of the local maximum values of the obtained y_n , written as k_n .

(7) Judge whether $k_n < k_b$ (k_b is the boundary value, in default $k_b = 30$). If no, n = n + 1 and repeat steps (5–7) until $k_n < k_b$. If yes, end the envelop interpolation process and output all the local maximum values that are stored in the matrix $N_1(f, A)_{kn \times 2}$, where f is the potential instantaneous frequency, A is the amplitude value of each local maximum, and k_n is the initial maximum iteration number.

$$N_1(f,A) = \begin{bmatrix} f_1 & A_1 \\ \dots & \dots \\ f_{kn} & A_{kn} \end{bmatrix}$$
(13)

Subsequent work in this section is further carried out to explain the reduction in the local maximum values through the simulated signal described in Section 2.3. The local maximum values in the spectrum of the original signal are represented by an inverted red triangle in Figure 7, with a total amount of 1330. However, the analysis result is unreasonable and not ideal because the spectrum contains too many details. Therefore, there is a need to reduce the number of extreme points.



Figure 7. The local maximum values in the frequency spectrum of the original signal.

After calculating the envelope of $y_0(t)$, the spherical spline interpolation method is used to solve the problem caused by the envelope. This interpolation method makes it easy to smooth the envelope line and further reduce the number of extreme points. In order to confirm the effect of the envelope interpolation method clearly, the results after each execution are shown in Figure 8. The first execution result is shown in Figure 8a, where the original envelope lines are smoothed, showing that the minor peaks of the original data sequence were abandoned.



Figure 8. Envelope interpolation process: (**a**) 1st time execution; (**b**) 2nd time execution; (**c**) 3rd time execution; (**d**) 4th time execution; (**e**) 5th time execution.

As indicated in Figure 8b,c, after the second and third envelope and interpolation, the maximum values of the data sequence were greatly reduced. For the fourth time envelope displayed in Figure 8d, the detected local maximum values were reduced to 15. By applying this manipulation, the overall major local maximum values that stand for every local range could be recognized. However, if the number of executions was too large, the envelope line of the spectrum contained few local maximum values within a certain frequency range, or the envelope line ended up with a horizontal line.

In Figure 8e, only three local maximum values can be found within the fifth envelope line, and the important low-frequency maximum information between 0 and 1000 Hz was left out. Therefore, in order to avoid the loss of the typical global maximum values caused by too many iterations, a stop condition was set through k_b (in default $k_b = 30$). When the number of the local maximum values is less than k_b , the envelope interpolation process is stopped. After four envelope interpolations of the simulated signal, the number of local maximum values was $15(< k_b = 30)$, so the envelope stopped.

3.2.2. Adjustment

Even though the number of local maximum values decreased significantly through the multiple envelope and interpolation performance in Section 3.1, it would still be too redundant if each extremum point were iterated only once. As shown in the above envelope results of the simulating signals, using the ACMD algorithm to perform the iterative filtering for these 15 extreme points is cumbersome and unnecessary. Therefore, the correlation analysis method is employed to merge the extreme points with strong frequency correlation in $N_1(f, A)$, and the final iteration number and the instantaneous frequencies can be determined by the further reduced local maximum values. The updated K and f_c are stored in $N_2(f, A)$. The process of the further adjustment is shown in Figure 6 (see the green box) and the detailed instructions are as follows:

(1) Load the matrix $N_1(f, A)_{kn \times 2}$.

(2) Initialize the parameters $\alpha = 0.1$, i = 1, j = 1.

 $(3) f_{c1} = f_i, f_{c2} = f_{i+1}.$

(4) ACMD parameters (α , f_{c1}) and (α , f_{c2}) are set, then the original signal x is processed to obtain the filtered signals Z_1 and Z_2 .

(5) Calculate the Spearman rank correlation coefficient values of ρ for Z_1 and Z_2 , respectively. The spearman coefficient [45] is a non-parametric index of the statistical dependence between two observational stochastic sequences. It accesses the relationship among the sequences in which the correlation coefficient can be depicted by using a monotonic function as:

$$\rho = 1 - \frac{6\sum_{i=1}^{n} d_i^2}{q(q^2 - 1)} \tag{14}$$

where *d* is the difference between the sequences, and *q* is the number of sequences. The correlation coefficient $\rho > 0.6$ is generally considered to be strongly relevant.

(6) Judge whether $\rho > 0.6$. If yes, the components with the smaller amplitudes in the frequency domain are discarded, whereas the components with the larger amplitudes are stored in the matrix $N_2(f_j, A_j)$. Then j = j + 1, i = i + 1. If no, the information corresponding to the two frequency values is retained. $N_2(f_j, A_j) = N_1(f_i, A_i)$, $N_2(f_{j+1}, A_{j+1}) = N_1(f_{i+1}, A_{i+1})$, j = j + 2 and i = i + 1.

(7) Judge whether $i = k_n$. If yes, the maximum iteration number (*K*) is equal to the number of row vectors in the matrix $N_2(f, A)$, and then go to step 8. If no, repeat steps (3–7).

(8) The matrix $N_2(f, A)_{K \times 2}$ is rearranged in descending order of A so that the instantaneous frequency (f_c) of each iteration can be determined.

Section 3.2 is used to process the matrix $N_1(f, A)$ obtained in Section 3.1, and the 15 maximum values are reduced to 3. After calculation, the five maximum values in the blue area in Figure 9 are the strongly correlated points, so only the maximum value with the highest amplitude is retained in the matrix $N_2(f_1, A_1) = (400, 0.800)$. Similarly, through correlation analysis, the yellow region and the green region can be found as the other two related regions, and the final results of the mergers are $N_2(f_2, A_2) = (3000, 0.497)$, $N_2(f_3, A_3) = (5543, 0.061)$. The maximum iteration number (*K*) is 3. As $N_2(f, A)$ reordered in the descending order of *A*, the instantaneous frequencies of ACMD for each iteration can be also determined.



Figure 9. Adjustment process.

3.3. Weight Factor (α) Selection for ACMD

As mentioned, the selection of the weight factor (α) is also an important issue in ACMD. However, in rolling bearing signals, random impulses and harmonic components usually exist simultaneously. Consequently, different bandwidths should be provided to match each component. Additionally, in the actual signal, the energy of the random shocks, the harmonic components, and the noises is much more than that of the fault information. The aforementioned two points make it unreasonable to set a constant α to directly extract the fault information from the original signal or to progressively acquire the characteristic information of all the potential modes. Therefore, according to $N_2(f, A)_{K\times 2}$ obtained in Section 3.1, the complex components are separated step by step through the cyclic iteration. The optimization procedure of the parameter α for each component is shown in Figure 6 (see the blue box) and can be described as follows:

(1) Initialize the parameters $k_2 = 0.5$, $k_3 = 1$, $f = f_m$. where f_m is the instantaneous frequency obtained in Section 3.1.

(2) Calculate $\alpha_{k3} = 10^{-k2}$.

(3) The parameters f_c and α_{k3} of ACMD are set, and then the filtered signal Z_3 can be obtained.

(4) Calculate the *ELK* of Z_3 , denoted as L_{k3} .

(5) Judge whether $k_2 = 5$. If yes, go to step 6. If no, $k_2 = k_2 + 0.5$, $k_3 = k_3 + 1$, repeat steps (2–5).

(6) The α value corresponding to the maximum *ELK* is the optimal weight factor, which is denoted as α_m .

3.4. Feature Extraction Results of Each Iteration

The iteration output procedure is indicated in Figure 6 (see the dark blue dotted box) and can be summarized as follows:

(1) Load $N_2(f, A)_{K \times 2}$.

as

(2) Initialize the parameters $m = 1, f = f_m$.

(3) Initialize the input signal of the iteration $F_{in}(m, :) = x$.

(4) As described in Section 3.2, calculate the *ELK* values with different α values in $F_{in}(m, :)$. In this iteration, the optimal weight factor of ACMD is obtained at the largest *ELK*, denoted as α_m .

(5) ACMD parameters f_m and α_m are set, and $F_{in}(m, :)$ is processed to obtain the *m*-th filtered signal $F_{out}(m, :)$.

(6) Calculate the remanent signal $F_{in}(m + 1, :) = F_{in}(m, :) - F_{out}(m, :)$.

(7) Calculate the remanent energy ratio of $F_{in}(m + 1, :)$ to the original signal *x*, denoted

$$\varepsilon = \frac{\sum F_{\rm in}^2(m+1,:)}{\sum x^2} \tag{15}$$

(8) Judge whether $\varepsilon < 0.1\%$ and whether m = K. If yes, end the iteration process, and the filtered signals of *m* modes are stored in *F*_{out}. If no, m = m + 1 and repeat steps (4–8).

(9) Calculate the envelope spectra of the estimated m modes to determine whether the fault characteristic frequency of the bearing exists or not.

Similarly, taking the simulation signal as an example, *x* is further decomposed iteratively on the basis of $N_2(f, A)$. *ELK* values in different α cases for each F_{in} are calculated, as shown in Figure 10. Then the corresponding $\alpha(m)$ of each mode can be obtained.



Figure 10. *ELK* values with different α at each F_{in} .

The spectra of the three extracted target modes by ACMD with the instantaneous frequencies of 400 Hz, 3000 Hz, and 5543 Hz, respectively, are shown in the left column of Figure 11a. It is shown that the decomposition results were quite satisfactory because of the low correlation between each mode. The corresponding envelope spectra of the three extracted modes are displayed in the right column of Figure 11b. It can be found that the features of the different characteristic signals were successfully separated by the proposed method. The characteristic components of the rotating frequency (f_r), the harmonic interference, and the inner race fault information (f_{in}) are indicated in modes w_1 , w_2 , and w_3 , respectively. Based on the above analysis, it can be judged that the bearing had an inner race fault.



Figure 11. Analysis results of each mode of the simulated inner race fault signal: (**a**) frequency spectrum; (**b**) envelope spectrum.

4. Application Procedures of the Proposed Method

The entire flowchart of the proposed IMACMD method is displayed in Figure 12, and the specific steps to apply the method for bearing fault diagnosis are given below:





Step 1: Load the collected vibration signal *x*.

Step 2: Use the envelope interpolation method to reduce the number of local maximum values of the original signal in the frequency domain. All local maximum values are stored in the array $N_1(f, A)_{kn \times 2}$.

Step 3: Determine the maximum iteration number (*K*) and the instantaneous frequencies (f_c) of the ACMD algorithm by the correlation analysis method, denoted as $N_2(f, A)_{K \times 2}$.

Step 4: Initialize m = 1, $f = f_m$, and input signal $F_{in}(m, :) = x$.

Step 5: For $F_{in}(m, :)$, the optimal weight factor (α_m) of ACMD is obtained based on the *ELK* indicator.

Step 6: Based on f_m and α_m , the filtered signal $F_{out}(m, :)$ and the residual signal $F_{in}(m + 1, :)$ are obtained.

Step 7: Calculate the remanent energy ratio (ε) of $F_{in}(m + 1, :)$ to the original signal x. Step 8: Judge whether $\varepsilon < 0.1\%$ or whether m = K. If yes, end the iteration process and

store all the modes in F_{out} . Else, m = m + 1 and repeat steps (5–8).

Step 9: All the fault types are determined by the envelope spectrum of each mode.

5. Experimental Signal Verification

5.1. Experimental Setup Introduction

The effectiveness of the presented diagnostic frame was investigated by the vibration signals that were sampled from a QPZZ rotary machinery fault simulator. The details of the experimental fault simulator can be seen in Figure 13a. There was a loading device, an

eddy current sensor, a drive motor, and several bearing blocks. As shown in Figure 13d,e, two of the same SKF6025 type bearings with an outer race and an inner race defect were used in the experiments. Electrical discharge machining was employed to produce a tiny groove on the races of the bearing. During the vibration signal measurement, the sampling frequency was set to 12,800 Hz. The rotating speed was set to 1470 rpm, with the rotating frequency set as $f_r = 24.5$ Hz. According to the listed parameters of the SKF6205 bearings in Table 2, the theoretical fault characteristic frequencies of the inner race, the outer race, the roller, and the cage were calculated as f_{in} = 132.67 Hz, f_{out} = 87.83 Hz, f_{roller} = 115.48 Hz, and $f_{cage} = 9.76$ Hz, respectively.



(a)



(b)



(c)

Figure 13. The local machinery fault simulator: (a) The experimental fault simulation operation platform; (b) signal acquisition display; (c) acceleration sensor locations; (d) outer race fault bearing; (e) inner race fault bearing.

Table 2. Parameters of the SKF6205 bearing.

Ball Number	Ball Diameter	Pitch Diameter	Contact Angle
9	7.94 mm	39.04 mm	0°

5.2. Single Fault Signal Analysis and Comparison

A dataset of the original inner race fault signal with 8192 points, whose frequency resolution and local stability were satisfied, was selected for the single fault analysis. In Figure 14a, some periodic impacts can be seen from the waveform of the experimental signal. However, the interval between two adjacent impacts was about 0.04 s, which corresponds to the shaft-rotating period.



Figure 14. Experimental signal of bearing inner race single fault: (**a**) waveform; (**b**) frequency spectrum; (**c**) envelope spectrum.

The spectrum is shown in Figure 14b. It shows that the frequency components with high energy were mainly distributed below 3000 Hz, which means the resonance range was relatively wide. Moreover, the peak of 132.67 Hz, which was associated with the inner race defect, could not be observed. According to the envelope spectrum lines displayed in Figure 14c, no obvious spectral peaks could be found at the characteristic frequency of the inner race fault; see the red dotted line. On the contrary, the rotation frequency and its harmonics were obvious and in good accordance with the information in the time-domain waveform.

In the spectrum of the inner race fault signal, the number of local maximum values was 1304. As shown in Figure 15, the envelope interpolation method was performed four times on the original spectrum, and the local maximum values for each execution on the spectrum are marked by the red inverted triangles. It can be seen that the local maximum values after one, two, three, and four interpolations were 463, 156, 49, and 14, respectively.



Figure 15. The envelope interpolation method process of experimental inner race failure signal: (a) 1st time execution; (b) 2nd time execution; (c) 3rd time execution; (d) 4th time execution.

According to the ACMD filtering effect on the local extremum points of different frequencies, the maximum values with strong correlation to the filtering results were

reduced from 16 to 5, which is also the upper limit of the loop iteration. Therefore, the instantaneous frequencies of ACMD in each iteration were 661, 1886, 3259, 4313, and 5719, respectively. Based on the principle of the largest *ELK*, the optimal weight factor (α) of ACMD in each iteration was also captured, as shown in Figure 16a. Through the above optimization, the important parameters of the ACMD algorithm were set to complete the five time iterations. The inner race fault signal was decomposed into five modes, and the spectrum distribution is illustrated in Figure 16b. After processing by the IMACMD diagnosis method, the envelope spectrum of each mode is illustrated in Figure 16c, where the rotating frequency and its harmonics could be obviously found in modes w_1 and w_2 , whereas the inner race fault feature frequency and its harmonics could be clearly observed in mode w_4 .



Figure 16. Proposed diagnostic strategy analysis results of experimental inner race fault signal: (a) *ELK* values with different α at each F_{in} ; (b) frequency spectrum distribution results of the obtained modes; (c) envelope spectrums of the obtained modes.

For comparison purpose, the widely used methods of singular spectrum decomposition (SSD) and spectral kurtosis (SK) were also applied to the experimental single fault signal.

SSD requests the mode decomposition quantitative parameters be preset in advance. In order to make the comparison results convincing, the number of singular spectrum components (SSC) was also set to 5, which is the output mode number of IMACMD. In Figure 17, the analysis results of SSD show that the inner race fault component had a very low amplitude and was only included in the envelope spectrum of SSC₁. The rotating characteristics were widespread in all SSCs. These results show that SSD method was invalid for this experimental signal.



Figure 17. SSD decomposing results of the experimental inner race fault signal.

Then the experimental signal was processed by using the SK method. According to the SK kurtogram in Figure 18a, the optimal filtering frequency band was at level 5.5 and the center frequency was 2800 Hz. The envelope spectrum of the filtered signal by the constructed filter is displayed in Figure 18b. As indicated in Figure 18b, a lot of periodic impacts could be seen in the envelope spectrum, and their interval was about 24.5 Hz. Such an interval corresponds to the rotating frequency, but not the fault frequency.



Figure 18. SK filtering result of experimental inner race fault signal: (**a**) kurtogram; (**b**) envelope spectrum of the filtered signal.

Compared with the two widely used diagnosis methods, IMACMD obtained the most satisfying analysis results and had the advantage in the single bearing damage judgment.

5.3. Compound Fault Signal Analysis and Comparison

In this subsection, the compound race fault was tested to further validate the superiority of the proposed IMACMD method. In Figure 19a, the observable impacts (about 0.02 s) corresponding to the harmonic components were associated with the rotating frequency. Such impact could also be obviously seen in the envelope spectrum in Figure 19c at 24.5 Hz. In Figure 19b, the energy concentration of the spectrum ranged from 0 Hz to 2500 Hz. In the envelope spectrum, the blue dotted line and the red dotted line represent the theoretical frequencies of the outer and the inner race faults, respectively. However, the bearing fault characteristic components of neither the inner nor the outer race fault were detected in the spectrum or the envelope spectrum; see Figure 19b,c.



Figure 19. Experimental signal of bearing races compound fault: (**a**) waveform; (**b**) frequency spectrum; (**c**) envelope spectrum.

Then the proposed diagnostic method was also employed in the analysis of the experimental compound fault signal, whose analysis results are displayed in Figure 20. By using the four-time envelope interpolation method, the estimated local maximum values of the spectrum were gradually reduced from the initial 1313 to the final 13, as shown in Figure 20a. By analyzing the similarity of the ACMD filter signals at each extremum point, the maximum iteration number was locked at 5, and meanwhile the instantaneous frequency of each iteration was also obtained, which was 391, 2344, 4717, 3711, and 5809, respectively. However, after the fourth iteration, the residual signal took up less than 0.1% energy of the original signal. It can be considered that the key information of the

original signal had been extracted and the iteration ended. Therefore, through the dual constraints of the correlation analysis and the iteration end condition, the compound fault signal was eventually decomposed into four modes. According to Figure 20b, the optimal value of α for each iteration could be obtained. The spectrum distribution of each mode is illustrated in Figure 20c, whereas Figure 20d displays the envelope spectrums of the four separated modes. Modes w_1 and w_2 showed serial harmonics of f_r , whereas the envelope spectrum of mode w_3 revealed four harmonics of the inner race fault features. Particularly, the harmonics of the characteristic frequency $f_{out} \sim 6f_{out}$ could also be successfully detected through the envelope spectrum of w_4 . So far, the IMACMD method proposed in this paper successfully extracted and separated the inner and the outer race faults.



Figure 20. Proposed diagnostic strategy analysis results of experimental compound fault signal: (a) envelope and interpolation results; (b) *ELK* values with different α at each F_{in} ; (c) frequency spectrum distribution results of the obtained modes; (d) envelope spectrums of the obtained modes.

The number of SSCs was set to 4, and the compound fault signal analysis results of the SSD method are shown in Figure 21. Although the envelope spectra of SSC₁ and SSC₂ showed the fault characteristic frequency f_{out} and the envelope spectrum of SSC₃ displayed the fault characteristic frequency f_{in} , the amplitude of the fault characteristic frequency was still serious. The fault features of both the inner and the outer race were extracted insufficiently, which means that the SSD method was invalid for handling this experimental compound fault signal.



Figure 21. SSD decomposing results of the experimental compound fault signal.

The SK kurtogram for the compound fault signal analysis is displayed in Figure 22a, based on which the optimal frequency band was confirmed in level 5. Then the optimal filter with a center frequency of 6133.33 Hz was created to get the filtered signal. The corresponding envelope spectrum of the filtered signal is shown in Figure 22b. Obviously, it was easy to identify the harmonics of the rotating frequency at 24.5 Hz, whereas the fault characteristic frequencies of 132.67 Hz and 87.83 Hz could not be detected.

Therefore, for the composite fault signals, the acquired sensitive component using the IMACMD method could better separate the sensitive components and more effectively obtain the fault characteristics information than the SSD and SK methods.





6. Engineering Signal Verification

6.1. Wind Turbine Introduction

The proposed IMACMD diagnostic method was further validated through an actual bearing fault signal sampled from a 750 KW wind turbine generator. The schematic diagram of the wind turbine, which was composed of an impeller, a planetary gearbox, and a double-fed asynchronous generator, is illustrated in Figure 23a. The sensor positions are illustrated in Figure 23b, and the generator front–back stator casings are displayed in Figure 23.



Figure 23. Wind turbine: (a) schematic diagram; (b) accelerometers positions; (c) front-stator casing and back-stator casing.

Considering that the wind speed frequently varies in a random way, the rotating speed of the wind turbine consequently changes due to the mechanical torque/movement transfer. To validate the proposed method, we selected the data series that were sampled in a more stable wind speed condition, with a sampling frequency of 16,384 Hz. In the meantime, these long sampling data series were divided into several segments. Each segment was about 0.5 s, and within such short period the rotating speed of the wind turbine could be approximately treated as stable. The data segment employed for validation in this paper corresponded to a rotating speed of 1519 r/min.

The parameters of the defective SKF6324 bearing are listed in Table 3. In this practical engineering industrial equipment, local weak damage occurred on the outer race of the front generator bearing during the vibration monitoring. According to the structural details, the rotating frequency of the rotor $f_r = 25$ Hz. Furthermore, the theoretical characteristic

frequencies were also computed as f_{out} = 79.2 Hz, f_{in} = 123.2 Hz, f_{roller} = 110.9 Hz, and f_{cage} = 15.4 Hz, respectively.

Table 3. Parameters of the SKF6324 bearing.

Ball Number	Ball Diameter	Pitch Diameter	Contact Angle
8	41.275 mm	190 mm	0°

6.2. Engineering Signal Analysis and Comparison

As represented in Figure 24a, the impulsive phenomenon was unable to be observed from the waveform of this measured engineering signal with the data length of 8192 points. Figure 24b is the spectrum of the original signal, where there were only three significant frequency bands. Traditional envelope spectrum analysis was further performed on this vibration signal. In Figure 24c, the red dotted lines denote the fault frequency and its harmonics. Due to the harsh interference of the noise and the harmonics, only a rotating frequency of 25 Hz and its multiples were exhibited, whereas the fault characteristic frequency of the outer race could not be observed from the Hilbert envelope spectrum in Figure 24c. Thus, no diagnosis conclusion was able to be drawn based on the traditional analysis results.



Figure 24. Engineering fault signal: (a) waveform; (b) frequency spectrum; (c) envelope spectrum.

To detect the bearing fault, the IMACMD method was employed to analyze the engineering signal. After four envelope interpolations, the local maximum values in the spectrum of the original signal were reduced from 1327 to 13, as shown in Figure 25a.

From Spearman rank correlation analysis, the modes with strong similarity were combined, and finally four effective local maximum values were left. By sorting the amplitudes of these extreme points, the instantaneous frequencies of the four iterations were 1664 Hz, 4797 Hz, 3226 Hz, and 6254 Hz, respectively. These four modes were defined as the primary ones to constitute the main information of the engineering signal. For each mode, the optimal value of α and the spectrum distribution are shown in Figure 25b,c. As illustrated in Figure 24, the envelope spectra of mode w_1 - w_3 contained some harmonics, whereas the envelope spectrum of mode w_4 successfully separated the fault feature of the outer race.

To highlight the superiority of the proposed method, two different methods, i.e., SSD and SK, were also applied to analyze the actual engineering data for comparison.



Figure 25. Proposed diagnostic strategy analysis results of engineering fault signal: (**a**) envelope and interpolation results; (**b**) *ELK* values with different α at each F_{in} ; (**c**) spectrum distribution results of the obtained modes; (**d**) envelope spectrums of the obtained modes.

By the SSD method, the engineering fault signal was decomposed into four components, and the envelope spectra of SSCs are illustrated in Figure 26. The characteristic frequency of the outer race fault had a smaller amplitude s to that in w_4 in Figure 25d;

hence, it could only be picked up in SSC_1 . Meanwhile, a lot of harmonics of the rotating frequency can be seen in SSC_1 , SSC_2 , and SSC_3 .



Figure 26. SSD decomposing results of the engineering fault signal.

The comparison with SK is displayed in Figure 26. On the basis of the kurtogram information in Figure 27a, the optimal frequency bandwidth was at level 6 and the best center frequency was 6656 Hz. Then the envelope spectrum of the filtered signal could be presented, as shown in Figure 27b. It shows that only a feeble basic characteristic frequency was discovered, while its other harmonics were invisible. Obviously, the exhibited analysis of SK method was not as effective as that of IMACMD.



Figure 27. SK analysis results of the engineering fault signal: (**a**) kurtogram; (**b**) envelope spectrum of the filtered signal.

Through the previous engineering case analysis, an exact fault conclusion was unable to be drawn based on the analysis results of the SSD and SK methods. On the contrary, the proposed IMACMD method was more feasible and ascendant for weak fault feature extraction.

7. Conclusions

Strong noise removing and weak signal stripping from the vibration signal are crucial for bearing fault diagnosis, since in many cases the characteristics are often covered by heavy background noise. The ACMD method is a neoteric approach for adaptive signal processing and has a performance advantage in getting the target frequency characteristics. However, it is also greatly affected by some deficiencies. For instance, the detection of all the potential components and the key parameter setting needs prior knowledge. In this paper, a new iterative adaptive method, called IMACMD, is proposed to overcome these limitations. The contribution of this paper is as follows:

(1) As an iterative method, IMACMD can isolate all the modes from the original signal without the request of any prior knowledge about the rotation frequency, the fault frequency, etc. The proposed method is suitable for both single and compound fault signals, and the fault type can be identified successfully by the separated modes.

(2) The main advantage of IMACMD is using the dual constraints of the envelope interpolation as well as the Spearman rank correlation analysis to realize the minimization of the decomposition modes. Meanwhile, the instantaneous frequency parameters of each mode can be obtained.

(3) A new index named *ELK* is proposed in this paper, which is particularly effective for evaluating the degree of the periodicity from heavy noisy signals. Based on *ELK* index, the weight factor can be adaptively determined while extracting different mode components.

In the real datasets from experimental and engineering signals, the proposed method is superior to the SSD and SK methods in fault frequency detection, even though there is a variety of interferences from random impulses, large period fluctuations, and heavy noises. In this prospect, the proposed fault diagnosis method for rolling bearings has high potential to be widely applied, especially in multiple-fault and variable-speed cases.

In engineering, our method is qualified to diagnose both single and compound faults of bearings in approximately stable running conditions. However, in intensively varying speed cases, the proposed method is hard to use accurately. This issue will be investigated in future work.

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Abbreviations

ACMD	adaptive chirp mode decomposition
CYCBD	cyclostationary blind deconvolution
EMD	empirical mode decomposition
IMACMD	iterative modified adaptive chirp mode decomposition
LMD	local mean decomposition
MCKD	maximum correlated kurtosis deconvolution
MED	minimum entropy deconvolution
MOEDA	multipoint optimal minimum entropy deconvolution adjusted
SK	spectral kurtosis
SSD	singular spectrum decomposition
SSC	singular spectrum component
VMD	variational mode decomposition
Notations	1
Α	amplitude value of each local maximum
A_n	amplitude of cycle impacts
B_i	amplitude of <i>j</i> th random impulse
C_k	amplitude of <i>k</i> th harmonic component
d	difference between the sequences
ε	remanent energy ratio
f	potential instantaneous frequency
fc	instantaneous frequency
f_{cage}	cage fault characteristic frequency
$f_{\rm in}$	inner race fault characteristic frequency
f_k	resonance frequency of <i>k</i> th harmonic component
f_m	coefficient of resonance frequency
f_n	resonance frequency of cycle impacts
fout	outer race fault characteristic frequency
$f_{\mathbf{r}}$	rotating frequency
f_{roller}	roller fault characteristic frequency
f_s	sampling frequency
f_v	resonance frequency of random impulses
Ι	number of cycle impacts
J	number of random impulses
Κ	maximum iteration number
k_b	boundary value of the local maximum
K _c	number of harmonic components
k_n	number of local maximum values
N	sampling points
9	number of sequences
s _m	impulse response function of the rotating machinery system
t	time
T_a	period of cycle impacts
T_{j}	occurrence time of <i>j</i> th random impulse
w	mode of IMACMD
α	weight factor
β_m	coefficient of resonance damping
γ	specifies the slippage characteristic
μ _r	rth L-moment
ρ	Spearman rank correlation coefficient values
φ_k	phase of <i>k</i> th harmonic component
φ_m	coefficient of phase

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